

Astrophysical Fluid Dynamics

(1)

What are fluids? • things that can flow

- eg liquids (only on planet surfaces/interiors, incompressible)
- gases (usually compressible in astrophysics)
- Not solids

Key concepts

- ignore individual particles
- treat fluid as continuous medium made of large # of particles
- at each point fluid has well-defined macroscopic properties (eg. density, pressure)

Fluid element

- fluid approximation requires that it's possible to define a region of size L in which local variables (eg ρ, p) can be defined
- $\therefore L \ll q / |\nabla q|$ ie. small enough to ignore systematic variation in variable q
- $nL^3 \gg 1$ ie. large enough to ignore finite particle number (discreteness noise) ($n = \# / \text{volume}$)

Collisional fluid

- also $L \gg \lambda = \text{mean free path}$, so that particles know about local conditions
- If particles interact, fluid attains a distribution of particle speeds that maximises entropy \rightarrow that distr. is well defined for given T, ρ
- Can define an equation of state, $p(T, \rho)$, that includes all the microphysical complexity
- Inverse fluids considered to same hydrodynamic eqns, but different e.o.s.

Collisionless fluid

- distribution of speeds depends on initial conditions
- eg stars in galaxies, particles in Saturn's rings.

Two approaches to formulating fluid equations

Eulerian

- considers volume at fixed posⁿ Σ and specifies fluid properties as a fⁿ of time in that volume, eg $\rho(\Sigma, t), T(\Sigma, t)$
- rate of change at that posⁿ is, eg, $\partial \rho / \partial t$
- eg viewing a river from river bank
- eg solve numerically using grid-based code such as AMR (Adaptive Mesh Refinement)

Lagrangian

- considers a fluid element a (eg coords at $t=0$) that comoves w fluid, and specifies fluid properties as a fⁿ of time in that element, eg $\rho(a, t)$
- rate of change in that element is, eg, $D\rho/Dt$
- eg viewing a river from a boat adrift on it
- eg. solve numerically using ensemble of particles representing fluid elements and following their trajectories, such as SPH (Smoothed Particle Hydrodynamics)

Steady flows

quantities at given posⁿ don't change $\therefore \partial / \partial t = 0$ everywhere \rightarrow use Eulerian

Derivative following fluid motion

- Consider any quantity Q (scalar or vector) in fluid element at posⁿ \underline{r} at time t
- At time $t + \delta t$ the element is at $\underline{r} + \delta \underline{r}$, so:

$$DQ/Dt = \lim_{\delta t \rightarrow 0} \left[\frac{Q(\underline{r} + \delta \underline{r}, t + \delta t) - Q(\underline{r}, t)}{\delta t} \right]$$

$$= \lim_{\delta t \rightarrow 0} \frac{1}{\delta t} \left[\underbrace{Q(\underline{r}, t + \delta t) - Q(\underline{r}, t)}_{\text{Eulerian time derivative}} + \underbrace{Q(\underline{r} + \delta \underline{r}, t + \delta t) - Q(\underline{r}, t + \delta t)}_{\text{Convective derivative (because element has moved)}} \right]$$

Taylor expansion to 1st order in $\delta \underline{r}$ and δt :

$$= \lim_{\delta t \rightarrow 0} \frac{1}{\delta t} \left[\delta t \frac{\partial Q(\underline{r}, t)}{\partial t} + \delta \underline{r} \cdot \nabla Q(\underline{r}, t + \delta t) \right]$$

0 to 1st order

$$\nabla Q(\underline{r}, t) + \delta t \frac{\partial \nabla Q}{\partial t} + \dots$$

- As fluid velocity $\underline{u} = \delta \underline{r} / \delta t$

$$DQ/Dt = \underbrace{\frac{\partial Q}{\partial t}}_{\text{Eulerian time derivative}} + \underbrace{\underline{u} \cdot \nabla Q}_{\text{Convective derivative (because element has moved)}}$$

Convective derivative

Scalar Q : $\underline{u} \cdot \nabla Q = u_x \frac{\partial Q}{\partial x} + u_y \frac{\partial Q}{\partial y} + u_z \frac{\partial Q}{\partial z}$ in Cartesian coords

$= u_i \partial_i Q$ using summation convention (sum over subscripts that appear twice)
 $\partial_i \equiv \partial/\partial x_i$ where x_i is i th coord. variable

or $= u_r \frac{\partial Q}{\partial r} + \frac{u_\theta}{r} \frac{\partial Q}{\partial \theta} + \frac{u_\phi}{r \sin \theta} \frac{\partial Q}{\partial \phi}$ in Spherical polars

or $= u_R \frac{\partial Q}{\partial R} + u_z \frac{\partial Q}{\partial z} + \frac{u_\phi}{R} \frac{\partial Q}{\partial \phi}$ in cylindrical polars

Vector $\underline{Q} = [Q_x, Q_y, Q_z]$:

$$\underline{u} \cdot \nabla \underline{Q} = [u_x \frac{\partial Q_x}{\partial x} + u_y \frac{\partial Q_x}{\partial y} + u_z \frac{\partial Q_x}{\partial z}, u_x \frac{\partial Q_y}{\partial x} + \dots]$$

$$= u_i \partial_i Q_j$$

Kinematics = study of particle trajectories

• Consider velocity field defined everywhere in Eulerian coords $\underline{u}(\underline{r}, t)$

① Streamlines - dir of local velocity is tangent to streamline i.e. $dy/dx = u_y/u_x$

- defined by $(dt =) \frac{dx}{u_x} = \frac{dy}{u_y} = \frac{dz}{u_z}$

- for spherical polars $\underline{u} = (r, r\dot{\theta}, r\sin\theta\dot{\phi})$ so $(dt =) \frac{dr}{u_r} = \frac{r d\theta}{u_\theta} = \frac{r\sin\theta d\phi}{u_\phi}$

- for cylindrical polars $\frac{dr}{u_r} = \frac{dz}{u_z} = \frac{R d\phi}{u_\phi}$

② Particle paths - particle motion instantaneously follows streamline $d\underline{r}/dt = \underline{u}(\underline{r}, t)$

- but not identical if flow is time varying

- $\dot{x} = u_x, \dot{y} = u_y, \dot{z} = u_z$ cart.

- $\dot{r} = u_r, r\dot{\theta} = u_\theta, r\sin\theta\dot{\phi} = u_\phi$ sph pol.

- $\dot{R} = u_r, \dot{z} = u_z, R\dot{\phi} = u_\phi$ cyl. pol.

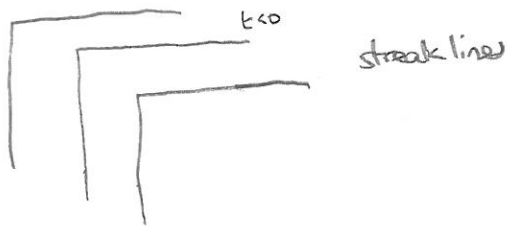
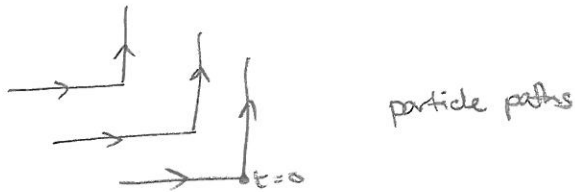
③ Streaklines - lines joining all particles which have ever passed through a point \underline{r}_0

- think of adding dye to particles as they pass

- find \underline{a} for which $\underline{r}(\underline{a}, t) = \underline{r}_0$ then streakline is $\underline{r}(\underline{a}, 0)$

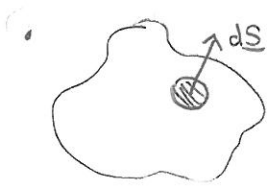
• For steady flows ① = ② = ③

• If, say, $\underline{u} = (1, 0, 0)$ for $t < 0$ and $\underline{u} = (0, 1, 0)$ for $t > 0$ we get



Fluid equations: conservation of mass (rate of change of mass in region = net inflow of mass)
 " momentum (" momentum " is balanced by momentum flow and net force)
 " energy (" energy " is determined by external energy gains/losses)

Conservation of mass



Consider a fixed volume V whose surface S is patchwork of surface elements $d\mathbf{S}$ = vector normal (outward) to surface, $|d\mathbf{S}|$ = area

Let ρ = mass density of fluid

\therefore Rate of change of mass in V is $\frac{\partial}{\partial t} \int_V \rho dV$

• Outward mass flow across $d\mathbf{S}$ is $\rho \mathbf{u} \cdot d\mathbf{S}$ = density \times length travelled in dir = $d\mathbf{S}$ per unit time \times area

Integrating over surface, rate of mass gain is $\oint_S \rho \mathbf{u} \cdot d\mathbf{S}$ - as $d\mathbf{S}$ is outward

By divergence theorem is $-\int_V \nabla \cdot (\rho \mathbf{u}) dV$

• If no sources/sinks of mass $\frac{\partial}{\partial t} \int_V \rho dV = -\int_V \nabla \cdot (\rho \mathbf{u}) dV$

As true for all volumes $\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0 \rightarrow$ Eulerian form of continuity eq.

• In Lagrangian form $D\rho/Dt = \partial\rho/\partial t + \mathbf{u} \cdot \nabla\rho$

$D\rho/Dt + \nabla \cdot (\rho \mathbf{u}) - \mathbf{u} \cdot \nabla\rho = 0$

$D\rho/Dt + \rho \nabla \cdot \mathbf{u} = 0$ $\Leftrightarrow \nabla \cdot (\rho \mathbf{u}) = \rho \nabla \cdot \mathbf{u} + \mathbf{u} \cdot \nabla\rho$

Incompressible $\equiv D\rho/Dt = 0$
 $\nabla \cdot \mathbf{u} = 0$

Pressure

Consider any surface element $d\mathbf{S}$ in fluid

• Pressure is the one-sided momentum flux associated w random motion of particles w.r.t. local rest frame of fluid due to their finite temperature

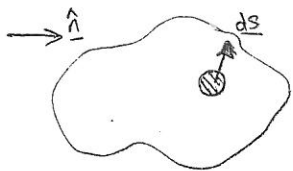
[balanced by = and opposite flux through other side of $d\mathbf{S}$]

• If random motion isotropic: mom. flux is indep. of orientation of $d\mathbf{S}$ and components \parallel to $d\mathbf{S}$ cancel

\therefore pressure force acting on $d\mathbf{S}$ is defined as $-\rho d\mathbf{S}$

(ie p is force/unit area)

Inviscid momentum equations ignoring viscosity



- Consider fluid element subject to gravitational accelerⁿ g and pressure from surrounding fluid
- Consider components of forces in direction \hat{n}

• Integrate pressure force over surface $F_n = \oint_S -p \hat{n} \cdot d\mathbf{S}$

$$= -\int_V \nabla \cdot (p \hat{n}) dV \quad \text{from divergence}$$

$$= -\int_V \hat{n} \cdot \nabla p + p \nabla \cdot \hat{n} dV \quad \text{from identity}$$

• Equation of motion for fluid element in \hat{n} dirⁿ is:

$$\left(\frac{D}{Dt} \left[\int_V \rho \underline{u} dV \right] \right) \cdot \hat{n} = -\int_V \hat{n} \cdot \nabla p dV + \int_V \rho \underline{g} \cdot \hat{n} dV$$

• For small volume $\int_V dV \rightarrow \delta V$ as local variables constant across vol.

$$\left(\underline{u} \frac{D}{Dt} [\rho \delta V] + \rho \delta V \frac{D\underline{u}}{Dt} \right) \cdot \hat{n} = -\nabla p \delta V \cdot \hat{n} + \rho \delta V \underline{g} \cdot \hat{n}$$

• As holds for all \hat{n} : $\underline{g} \frac{D\underline{u}}{Dt} = -\nabla p + \rho \underline{g}$ in Lagrangian form
 → momentum of fluid element changes due to pressure gradients and grav. forces

$\rho \frac{\partial \underline{u}}{\partial t} + \rho \underline{u} \cdot \nabla \underline{u} = -\nabla p + \rho \underline{g}$ in Eulerian form
 → mom in grid cell changes due to pressure gradient and grav forces, and net influx of mom into cell.

• Consider ^(Eulerian) rate of change of momentum density ($\rho \underline{u}$)

$$\frac{\partial \rho \underline{u}}{\partial t} = \rho \frac{\partial \underline{u}}{\partial t} + \underline{u} \frac{\partial \rho}{\partial t}$$

$$= -\nabla p + \rho \underline{g} - \underbrace{\rho \underline{u} \cdot \nabla \underline{u}}_{\text{from mom}} - \underbrace{\underline{u} \cdot \nabla (\rho \underline{u})}_{\text{from mass}}$$

• In Cartesian coords, using summation convention, the component in i th dirⁿ is

$$\frac{\partial \rho u_i}{\partial t} = -\partial_j p \delta_{ij} + \rho g_i - \rho u_j \partial_j u_i - u_i \partial_j \rho u_j$$

where δ_{ij} = Kronecker delta = 1 if $i=j$ 0 if not

NB $\underline{u} \cdot \nabla \underline{u} = [u_x \partial u_x / \partial x + u_y \partial u_x / \partial y + u_z \partial u_x / \partial z, \dots]$ eg see hand-out

$$\underline{u} \cdot \nabla (\rho \underline{u}) = [u_x (\partial \rho u_x / \partial x + \partial \rho u_y / \partial y + \partial \rho u_z / \partial z), u_y (\partial \rho u_x / \partial x + \dots)]$$

• Last two terms combine to $-\partial_j \rho u_i u_j$

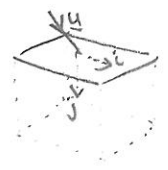
$$\therefore \frac{\partial \rho u_i}{\partial t} = -\partial_j \sigma_{ij} + \rho g_i$$

where $\sigma_{ij} = \rho u_j u_i + p \delta_{ij}$ is Stress tensor

• More generally, σ_{ij} is the force per unit area in dir = i acting on a surface whose normal is in dir = j

- thermal pressure, $p\delta_{ij}$, associated w random motions \tilde{u} which are isotropic (diagonal elements, p is scalar)

- ram pressure, $\rho u_i u_i$, associated w bulk motion u (3×3 matrix)



ρu_j = mass hitting surface w normal in j dir per unit time due to bulk flow

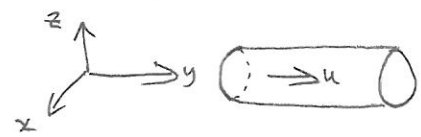
$\rho u_j u_i$ = momentum flux per unit time on that surface in the i dir

$\partial_j(\rho u_j u_i)$ = change in i th component of momentum due to mismatch in "i-momentum" carried through a cell in each of 3 orthogonal dirs

e.g. in x dir = $\frac{\partial}{\partial x}(\rho u_x^2) + \frac{\partial}{\partial y}(\rho u_x u_y) + \frac{\partial}{\partial z}(\rho u_x u_z)$

= other any microscopic effects giving such a stress can be included as a pressure (eg. mag. fields).

EG: flow down a pipe



All surfaces experience momentum flux due to thermal pressure

But only surfaces normal to flow experience ram pressure

$$\sigma_{ij} = \begin{pmatrix} p & 0 & 0 \\ 0 & p + \rho u^2 & 0 \\ 0 & 0 & p \end{pmatrix}$$

∴ Pressure on sides of pipe is p

end of pipe is $p + \rho u^2$

Gravity

- Is a conservative force \rightarrow work done on closed loop $\oint \underline{F} \cdot d\underline{l} = 0$
- By Stokes' theorem $\int_S \nabla \wedge \underline{F} \cdot d\underline{S} = 0$
- As "curl of grad is zero", conservative forces can be written $\underline{F} = \nabla \Phi$ where Φ is a scalar potential whose gradient gives magnitude of force
- Define gravitational potential Ψ s.t. grav. accel. $\underline{g} = -\nabla \Psi$ (-ve so that force acts in dir. of diminishing potential)
- Consider work required to take unit mass from r to ∞ where $\underline{g}(\infty) \rightarrow 0$ as far from all masses

$$-\int_r^\infty \underline{g} \cdot d\underline{l} = \int_r^\infty \nabla \Psi \cdot d\underline{l} = \Psi(\infty) - \Psi(r)$$
 independent of the path taken.
 Note that only potential differences or potential gradients are important

• Ex Let $\Psi = -GM/r$

$$\underline{g} = -\nabla \Psi = GM \nabla \left(\frac{1}{r}\right)$$

$$= -\frac{GM}{r^2} [\partial r / \partial x, \partial r / \partial y, \partial r / \partial z]$$

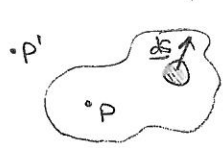
As $r^2 = x^2 + y^2 + z^2$, $\partial r / \partial x = x/r$ etc

$$\therefore \underline{g} = -\frac{GM}{r^3} [x, y, z] = -\frac{GM}{r^2} \hat{r}$$

So given potential is that of a point mass M at origin

- For a point mass M at \underline{r}' : $\Psi = -GM / |\underline{r} - \underline{r}'|$
- For multiple point masses M_i each at \underline{r}'_i , since ∇ is a linear operator summing potentials \rightarrow summing accelerations, so $\Psi = -\sum_i GM_i / |\underline{r} - \underline{r}'_i|$

Poisson's equation



- Consider a volume V bounded by surface S made of surface elements dS
- Consider a point mass M at P that exerts grav. acc. $\underline{g} = -\frac{GM}{r^2} \hat{r}$
- The quantity $\underline{g} \cdot d\underline{S} = -GM(d\underline{S} \cdot \hat{r}/r^2) = -GMd\Omega$
where $d\Omega = d\underline{S} \cdot \hat{r}/r^2$ is solid angle subtended by $d\underline{S}$ at P
- Integrate over surface $\int_S \underline{g} \cdot d\underline{S} = -GM\Omega$



- where $\Omega = \int_S d\Omega$ depends on whether P is inside or outside surface
- If P outside S : $\Omega = \int_S \hat{r} \cdot d\underline{S}/r^2 = 0$ because every l.o.s. cuts surface twice with equal and opposite contributions
 $= \int_V \nabla \cdot (\hat{r}/r^2) dV = 0$ alternatively use divergence theorem and use handout to show this is zero
- If P inside S , can't apply divergence theorem at P as singular, so split volume into small sphere around P of radius a , and the rest which has zero contribution by above arguments

$$\Omega = 4\pi a^2/a^2 = 4\pi$$

- So $\int_S \underline{g} \cdot d\underline{S} = -4\pi GM$ if P is inside S , 0 if P is outside
- If there is mass distributed through space, it is only that within S that matters
- $\int_S \underline{g} \cdot d\underline{S} = -4\pi G \int_V \rho dV = -4\pi G \times \text{mass enclosed by Gaussian surface}$

But $\int_S \underline{g} \cdot d\underline{S} = \int_V \nabla \cdot \underline{g} dV$ (divergence theorem)

$$\int_V \nabla \cdot \underline{g} + 4\pi G \rho dV = 0$$

$$\nabla \cdot (-\nabla \Psi) + 4\pi G \rho = 0 \quad (\text{as holds for all } V)$$

$$\underline{\nabla^2 \Psi = 4\pi G \rho}$$

* $\nabla \cdot (A_r \hat{r}) = \frac{1}{r^2} \frac{\partial}{\partial r} (r^2 A_r)$

EC: Spherically symmetric mass distⁿ $\rho(r)$

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Choose Gaussian surface as sphere radius r



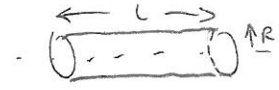
By symmetry \underline{g} is radial w same magnitude everywhere on surface $-|g|\hat{r}$

$$\therefore \int_S \underline{g} \cdot d\underline{S} = -4\pi r^2 |g| \quad \text{and this} = -4\pi G \int_0^r 4\pi \rho(r') r'^2 dr'$$

$$\therefore |g| = \frac{GM(r)}{r^2} = \frac{G}{r^2} \int_0^r 4\pi \rho(r') r'^2 dr' \quad \text{where } M(r) = \text{enclosed mass}$$

EC: Infinite cylindrically symmetric mass distⁿ $\rho(R)$

Choose Gaussian surface as cylinder radius R , length L



By symmetry \underline{g} is radial $= -|g|\hat{R}$ on surface of cylinder

As $\underline{g} \cdot d\underline{S}$ is zero on flat ends $\int_S \underline{g} \cdot d\underline{S} = -2\pi R L |g|$

$$\text{and this} = -4\pi G \int_0^R 2\pi R' L \rho(r') dr'$$

$$\therefore |g| = \frac{2G}{R} \int_0^R 2\pi R' \rho(r') dr'$$

\rightarrow depends only on interior mass distribution

EC: Infinite planar distribution symmetric about $z=0$, $\rho(z)$

Choose Gaussian surface as box of area A height $2z$



By symmetry \underline{g} is vertical $= -|g|\hat{z}$ on top and bottom

and $\underline{g} \cdot d\underline{S}$ is zero on sides $\therefore \int_S \underline{g} \cdot d\underline{S} = -2A|g|$

$$\text{and this} = -4\pi G A \int_{-z}^z \rho(z') dz'$$

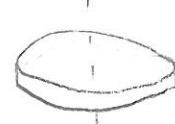
$$\therefore |g| = 4\pi G \int_0^z \rho(z') dz'$$

Note that $|g|$ is constant w height above z_{max} if $\rho(z > z_{max}) = 0$

EC: finite axisymmetric disk about $z=0$, $\rho(R, z)$

No surfaces where $\underline{g} \cdot d\underline{S}$ vanishes by symmetry

g.



$\therefore \underline{g}$ locally does not just depend on enclosed mass.

\rightarrow Galactic dynamics course

Spherically symmetric potential

$$\underline{g} = -\left(\frac{G}{r^2} \int_0^r 4\pi r'^2 \rho(r') dr'\right) \hat{r} = -\nabla \Phi \stackrel{\text{from hand-out}}{=} -(\partial \Phi / \partial r) \hat{r} = -(d\Phi/dr) \hat{r}$$

$$\therefore \int_{\infty}^r d\Phi = \int_{\infty}^r \frac{G}{r'^2} \int_0^{r''} 4\pi r'^2 \rho(r') dr' dr''$$

Set $\Phi(\infty) = 0$ and integrate by parts

$$\begin{aligned} \Phi(r) &= \left[-\frac{G}{r''} \int_0^{r''} 4\pi r'^2 \rho(r') dr' \right]_{\infty}^r + \int_{\infty}^r 4\pi G r'' \rho(r'') dr'' \\ &= \frac{GM(r)}{r} - \frac{GM(r)}{r} + \int_{\infty}^r 4\pi G r'' \rho(r'') dr'' \end{aligned}$$

So, while mass outside r does not affect acceleration at r , it does affect potential at r , since we set $\Phi(\infty) = 0$ and adding more mass beyond r means more energy req. to move particle to ∞ .

Gravitational potential energy

For a system of point masses, grav potential at \underline{r} is $\Psi = -\sum_i \frac{GM_i}{|\underline{r} - \underline{r}_i|}$

which is also energy req. to take unit mass from \underline{r} to ∞

\therefore Grav. pot. en. $\Omega = -$ energy req. to take all M_j to ∞

$$= -\frac{1}{2} \sum_{j \neq i} \sum_i \frac{GM_i M_j}{|\underline{r}_j - \underline{r}_i|}$$

only count each pair once energy to take M_j to ∞

$$= \frac{1}{2} \sum_j M_j \Psi_j \leftarrow \text{potential at } \underline{r}_j$$

So for distributed masses $= \frac{1}{2} \int \rho(\underline{r}) \Psi(\underline{r}) dV$

If spherically symmetric $= \frac{1}{2} \int_0^\infty 4\pi r'^2 \rho(r') \Psi(r') dr'$

Integrate by parts $= \frac{1}{2} \left[\int_0^\infty M(r') \Psi(r') \right]_0^\infty - \int_0^\infty M(r') \frac{d\Psi}{dr'} dr'$

But $d\Psi/dr = \frac{G}{r^2} M(r) \therefore \Omega = -\frac{1}{2} \int_0^\infty \frac{GM(r)^2}{r^2} dr'$

Integrate by parts $= -\frac{1}{2} \left[\int_0^\infty \left[-\frac{G}{r'} M(r')^2 \right]_0^\infty + \int_0^\infty \frac{G}{r'} 2M(r') \frac{dM}{dr'} dr' \right]$

$$= -G \int_0^\infty \frac{M(r')}{r'} dM(r')$$

(as expected as PE of spherical shell is $dM(r') \times \frac{GM(r')}{r'}$)

Moment of inertia

- Consider motion of cloud of particles (atoms / stars / galaxies)
- If a particle of mass m at \underline{r} is acted on by force $\underline{F} \rightarrow m\ddot{\underline{r}} = \underline{F}$
- Consider 2nd derivative of moment of inertia of particle about origin

$$\begin{aligned} \frac{1}{2} \frac{d^2(mr^2)}{dt^2} &= m \frac{d}{dt}(\underline{r} \cdot d\underline{r}/dt) = m\underline{r} \cdot \ddot{\underline{r}} + m\dot{\underline{r}}^2 \\ &= m\dot{\underline{r}}^2 + \underline{r} \cdot \underline{F} \end{aligned}$$

• Sum over all particles

$$\frac{1}{2} \frac{d^2 I}{dt^2} = 2T + \sum \underline{r} \cdot \underline{F}$$

where $I = \sum mr^2$, $T = \text{total kinetic energy}$, $V = \sum \underline{r} \cdot \underline{F} = \text{Virial}$

Virial

- If \underline{F}_{ij} is force on m_j from m_i then $\underline{F}_{ij} = -\underline{F}_{ji}$
 - ∴ this pair contributes $\underline{F}_{ij} \cdot \underline{r}_{ij}$ to V where $\underline{r}_{ij} = \underline{r}_i - \underline{r}_j$ to avoid double-counting
 - ∴ in absence of external force field $V = \frac{1}{2} \sum_{j \neq i} \sum_i \underline{F}_{ij} \cdot \underline{r}_{ij}$
- If ideal gas laws apply s.t. coll = processes occur at $\underline{r}_i = \underline{r}_j$, only gravity matters for which $\underline{F}_{ij} = -(Gm_i m_j / r_{ij}^3) \underline{r}_{ij}$
 - ∴ $V = -\frac{1}{2} \sum_{j \neq i} \sum_i Gm_i m_j / r_{ij}$
 - $= -\Omega$ grav. pot. en. of cloud

Virial theorem

In steady state I is constant.

$$\therefore 2T + \Omega = 0$$

Note that may need to decompose T (total kinetic energy) into that of the bulk flow (T_k) and that stored in particle motions in rest frame of fluid element, i.e. the thermal, or internal, energy (\mathcal{E}).

Equation of state

- To solve mom. eq., need to know pressure in fluid.
- e.o.s. is relⁿ betw. p and other thermodynamic properties that can be defined for collisional fluid

Ideal gases

- Most fluids in astronomy are approx. ideal
 - at microscopic level well described by kinetic theory where point-like particles have isotropic random motions set by fluid temperature
 - particles interact, but infrequently s.t. internal energy is (mostly) in kinetic energy (rather than PE. of interparticle interactions)

→ internal energy / unit mass $E = E(T)$ (only a fⁿ of temperature)

→ e.o.s: $p = \frac{R_*}{\mu} \rho T$

where $R_* =$ modified gas constant $= 8300 \text{ J/K/kg} = k/m_H$ Boltzmann's const
 $\mu =$ mean molecular weight Mass of H atom
 NB $p = nKT$ s.t. $\rho = n\mu m_H$ #/vol.

- Breaks down at high ρ or low T s.t. energy of interparticle interactions or finite particle size important and $E = E(\rho, T)$ and e.o.s. modified, e.g. giant planet interiors
- Also high ρ in neutron \star s, WDs \Rightarrow distⁿ of particle energies set by quantum mechanics \therefore ideal gas e.o.s. not appropriate

Barotropic e.o.s.

- Generally to get p need T from energy eqn^s that determine internal energy of fluid by considering heat gain/loss mechanisms.

But energy eqn^s not needed if e.o.s. is barotropic $\rightarrow p = p(\rho)$

(i) Isothermal e.o.s.

- T is const. $\therefore p \propto \rho$ (for ideal gas)
- Requires heating and cooling processes to control T to narrow range and for thermal \approx (heating balances cooling) to be attained on timescales shorter than flow timescales

(ii) Adiabatic e.o.s.

- Fluid elements that are thermally isolated from surroundings undergoing reversible changes have an e.o.s. $p = K\rho^\gamma$ where $K = \text{const}$
- NB this is different to an isentropic fluid in which same e.o.s. applies, but all elements in fluid have same K .

Thermodynamics review

Reversible changes = energy is conserved (ie., no viscous or dissipative processes converting kinetic energy into heat)

1st law of thermodynamics $dQ = dE + p dV$ for reversible changes (< if not)

heat absorbed from surroundings } change in internal energy } work done in changing vol by dV } per unit mass

NB d is Pfaffian operator meaning change dep. on route through thermodynamic phase space, and $V = 1/\rho$

For ideal gas $E = E(T) \therefore dE = \left(\frac{dE}{dT}\right) dT$

Define $C_x = (dQ/dT)_{x=const}$ = specific heat capacity at const. x

$\therefore C_v = dE/dT$

Subst. e.o.s. too

$dQ = C_v dT + \frac{R^*}{M} p T d\left(\frac{1}{p}\right)$

Adiabatic changes = no heat transfer across boundary $\therefore dQ = 0$

for an ideal gas undergoing a reversible adiabatic change:

$\frac{C_v}{R^*/M} \frac{dT}{T} = -\frac{dp}{p}$

and $p \propto T^{C_p/(R^*/M)}$, $p \propto T^{1 + C_v/(R^*/M)}$, $p \propto \rho^{1 - \frac{R^*/M}{C_v}}$

Rewrite using γ

ideal gas e.o.s $\Rightarrow p dV + V dp = \frac{R^*}{M} dT$

so reversible 1st law $\Rightarrow dQ = \underbrace{\left(C_v + \frac{R^*}{M}\right)}_{C_p} dT - V dp$

Defining $\gamma = C_p/C_v = 1 + (R^*/M)/C_v$

$\rho \propto T^{\frac{\gamma}{\gamma-1}}$, $p \propto T^{\frac{\gamma}{\gamma-1}}$, $p \propto \rho^\gamma$

What is γ ? depends on # of ways fluid can store kinetic energy

$C_v = N \frac{1}{2} (R^*/M)$ $\therefore \gamma = 1 + 2/N$

where $N = \#$ of degrees of freedom of fluid

eg. monatomic gas $N = 3$ (one for each orthogonal direction)

diatomic gas $N = 5$ (" + 2 rotational modes)

though depends on T as at low T fluid can be below threshold for excitation of particular rotation and vibration modes

Other ideal gas relations

$E = C_v T = \frac{1}{\gamma-1} \frac{p}{\rho}$

Energy equation

In absence of dissipative proc., 1st law of therm can be written

$$DE/Dt = DW/Dt + dQ/dt$$

rate of change of internal energy = work done - energy lost by cooling to surroundings (per unit mass)

where $DW/Dt = -p D(\frac{1}{\rho})/Dt = \frac{p}{\rho^2} D\rho/Dt$

Define a cooling function \dot{Q}_{cool} (per unit mass)

$$DE/Dt = \frac{p}{\rho^2} D\rho/Dt - \dot{Q}_{cool}$$

Define $E = \rho(\frac{1}{2}u^2 + \epsilon + \Psi)$ = total energy per unit volume
kinetic internal gravitational potential energy

$$DE/Dt = \frac{E}{\rho} D\rho/Dt + \rho(u \cdot Du/Dt + D\Psi/Dt + \frac{p}{\rho^2} D\rho/Dt - \dot{Q}_{cool})$$

- ①: $\partial E/\partial t + u \cdot \nabla E$ (by def of D/Dt)
- ②: $-\frac{E}{\rho} \rho \nabla \cdot u$ (by cons. of mass)
- ③: $u \cdot (-\nabla p - \rho \nabla \Psi)$ (by cons. of momentum)
- ④: $\rho \partial \Psi/\partial t + \rho u \cdot \nabla \Psi$ (by def of D/Dt)
- ⑤: $-\rho \frac{p}{\rho^2} \rho \nabla \cdot u = -\rho \nabla \cdot u$ (by cons. of mass)

$$\therefore \partial E/\partial t + \underbrace{u \cdot \nabla E + E \nabla \cdot u}_{\nabla \cdot (Eu)} + \underbrace{u \cdot \nabla p + p \nabla \cdot u}_{\nabla \cdot (pu)} = -\rho \dot{Q}_{cool} + \rho \partial \Psi/\partial t$$

$$\therefore \partial E/\partial t + \nabla \cdot ((E+p)u) = -\rho \dot{Q}_{cool} + \rho \partial \Psi/\partial t \text{ - usually } 0$$

Just need $E(p, T \text{ etc})$ and \dot{Q}_{cool}

Heating / cooling mechanisms

• Cosmic rays ^{high E particles (protons)} → free electrons → heat
↑
ionise atoms at rate / vol $\propto \rho \times$ cosmic ray flux

• Conduction particles' random motion allows internal energy to be transferred from hot to cold regions (by warm picles colliding w cooler ones)
heat flux / area = $-k \nabla T$
 k = thermal conductivity $\sim C_v (m k T / 3)^{1/2} / \sigma$

rate of change of energy / vol $\propto k \nabla^2 T$
(generally small, but equalises temperature & damps thermal instability, unless prevented by mag fields)
• Convection (eg. stars) is instability set up by gravity and temperature gradient
large scale fluid motion, but net effect can be modelled as transfer hot \rightarrow cold.

• Radiation energy carried by photons

- Optically thick emitted photons scattered and reabsorbed / scattered locally
diffusion problem is heat transfer hot \rightarrow cold until photons reach photosphere then escape to us

- Optically thin energy loss per unit vol per time by

+ recombination $n_e n_p k T \beta(H^0, T)$ (of free electrons onto H)
combina of recomb = cross section and vel. distrib

+ free-free rad (from e^- accelerated by picle of charge Z)
 $\propto Z^2 n_e n_p T^{1/2}$

+ collisionally excited atomic line rad e^- colls of atoms in ground state \rightarrow excite to low lying en. level, returning to ground \rightarrow emission of photons (energy)
 $\propto n_i n_e e^{-E/kT} \chi T^{-1/2}$
sometimes not enough en. to excite H (O^+, N^+)
determine from balance betw. ~~photo~~ ^{ionisation} and recombination
from * photons or colls

+ coll = betw molecules (KE \rightarrow rot^s or internal vibrs, important < 100K, O_2, O_3)

So... $\dot{Q}_{cool} = A \rho T^\alpha - H$
 ↑ ↑
 density enhancements precipitate cooling cosmic rays

Hydrostatic equilibrium

Static $\rightarrow \underline{u} = 0$ everywhere

Equilibrium $\rightarrow \partial/\partial t = 0$

Basic equations

Continuity: $\partial \rho / \partial t + \nabla \cdot (\rho \underline{u}) = 0$ trivial

Momentum: $\rho \partial \underline{u} / \partial t + \rho \underline{u} \cdot \nabla \underline{u} = -\nabla p - \rho \nabla \Phi$
 $\rightarrow \frac{1}{\rho} \nabla p = -\nabla \Phi$

Poisson's eq: $\nabla^2 \Phi = 4\pi G \rho$

Readily solved for barotropic eqn of state.

Eqn Isothermal slab

- Consider static isothermal slab, infinite in x and y and symmetrical about $z=0$. supported by gas pressure, under its own self gravity, no external forces

Isothermal $\rightarrow p = \left(\frac{R}{\mu} T\right) \rho = A \rho$

Geometry $\rightarrow \nabla = \partial/\partial z$ and $\rho = \rho(z)$, $\Phi = \Phi(z)$

\therefore Mooney $\Rightarrow \frac{1}{\rho} A d\rho/dz = -d\Phi/dz$

$\therefore A [\ln \rho]_{\rho_0}^{\rho} = [-\Phi]_{\Phi_0}^{\Phi}$ where $\rho_0 = \rho|_{z=0}$, $\Phi_0 = \Phi|_{z=0}$

$\therefore \Phi = \Phi_0 - A \ln(\rho/\rho_0)$

and $\rho = \rho_0 e^{-[(\Phi - \Phi_0)/A]}$

\therefore Poisson's eq $\Rightarrow d^2 \Phi / dz^2 = 4\pi G \rho_0 e^{-[(\Phi - \Phi_0)/A]}$

Let $\chi = -(\Phi - \Phi_0)/A$ and $Z = \sqrt{\frac{2\pi G \rho_0}{A}} z$

$\therefore d^2 \chi / dZ^2 = -2e^{\chi}$ w b.c. $\chi = d\chi/dZ = 0$ at $Z=0$

Multiply by $d\chi/dZ$ s.t. $\frac{d}{dZ} \left[\frac{1}{2} (d\chi/dZ)^2 \right] = \frac{d}{dZ} [-2e^{\chi}]$

Integrate to get $(d\chi/dZ)^2 = 4(1 - e^{\chi})$ from b.c.

$\therefore 2Z = I = \int_0^{\chi} (1 - e^{\chi})^{-1/2} d\chi$

Let $\chi = \ln \sin^2 \theta$ $I = \int_0^{\chi} \frac{2}{\sin(1-\sin^2)^{1/2}} d\chi$

$\chi = \ln \sin^2 \theta = \int_{\pi/2}^{\theta} \frac{2}{\sin \theta} d\theta$

$t = \tan \theta/2 = \int_1^t \frac{2}{t} dt = 2 \ln t$

$\therefore \rho = \rho_0 e^{\chi} = \rho_0 \sin^2 \theta = \rho_0 \frac{4t^2}{(1+t^2)^2} = \rho_0 \left[\frac{2}{e^{-Z} + e^Z} \right]^2 = \rho_0 / \cosh^2 \left[\sqrt{\frac{2\pi G \rho_0}{A}} z \right]$

Ex2 Isothermal atmosphere

Consider Earth's atmos. as plane w/ const $g = -g\hat{z}$

Isothermal $\rightarrow p = (\frac{R}{M}T)\rho$

Geometry $\rightarrow \nabla = \partial/\partial z$ and $\rho = \rho(z)$

Mon. eq $\rightarrow \frac{1}{\rho} (\frac{R}{M}T) d\rho/dz = -g$

$\rho = \rho_0 e^{-\frac{Mg}{R(T)}z}$ where $\rho_0 = \rho|_{z=0}$

As $\mu \approx 28, T \approx 300K$ then $\rho/\rho_0 \approx e^{-z/h}$ where $h = 9km$
 eg. at observatories at 5000m, density of atmos is ~60% that at sea level.

Ex3 Stars as self-gravitating polytropes

Consider spherical system (eg. rotate to break symmetry)

Geometry $\rightarrow \nabla = d/dr, p = p(r), \Phi = \Phi(r)$

Mon. eq $\rightarrow dp/dr = -\rho d\Phi/dr$

As $\rho > 0$, p is a monotonic function of Φ

$p = p(\Phi)$ (ie. unique value of Φ corresponds to unique value of p)

$dp/dr = (dp/d\Phi)(d\Phi/dr)$

$\rho = -(dp/d\Phi)$
 $= \rho(\Phi)$

Thus $p = p(\rho)$ and non-rotating stars are barotropes and surfaces of constant p, ρ and Φ coincide.

Polytropes parametrise barotropic e.o.s. as $p = K\rho^{1+\frac{1}{n}}$ where n = polytropic index

NB: This is only an approximation to $p(\rho)$ which is valid over some range of radii (and reasonable for entire interior)
 $\therefore 1 + \frac{1}{n}$ is not necessarily equal to $\gamma = C_p/C_v$, unless star was isentropic

Mon. eq $\rightarrow -d\Phi/dr = \frac{1}{\rho} dp/dr$
 $= \frac{1}{\rho} \frac{dp}{d\rho} \frac{d\rho}{dr} = \frac{K(n+1)}{n} \rho^{\frac{1}{n}-1} \frac{d\rho}{dr}$

Integrate from surface where $\rho = 0, \Phi = \Phi_T$

$\rho = [(\Phi_T - \Phi)/(n+1)K]^n$

If $\Phi = \Phi_c, \rho = \rho_c = [\frac{\Phi_T - \Phi_c}{(n+1)K}]^n$ at centre

$\rho = \rho_c \left[\frac{\Phi_T - \Phi}{\Phi_T - \Phi_c} \right]^n$

Poisson's eq $\rightarrow \nabla^2 \Phi = 4\pi G \rho_c \left[\frac{\Phi_T - \Phi}{\Phi_T - \Phi_c} \right]^n$

Let $\theta = \frac{(\Phi_T - \Phi)}{(\Phi_T - \Phi_c)}$ $\rightarrow \nabla^2 \theta = -\frac{4\pi G \rho_c}{\Phi_T - \Phi_c} \theta^n$

Now $\nabla^2 \theta = \frac{1}{r^2} \frac{d}{dr} (r^2 d\theta/dr)$

Let $\xi = \left[\frac{4\pi G \rho_c}{\Phi_T - \Phi_c} \right]^{1/2} r \rightarrow \frac{1}{\xi^2} \frac{d}{d\xi} \left[\xi^2 \frac{d\theta}{d\xi} \right] = -\theta^n$

This is Lane-Emden eqn of index n which has b.c. of $\theta = 1$ and $d\theta/d\xi = 0$ @ $\xi = 0$
 (ie., zero gravitational acceler. at centre, & unless there's a point mass at centre)

Solutions to the Lane-Emden equation $(\frac{1}{\xi^2} \frac{d}{d\xi} (\xi^2 \frac{d\theta}{d\xi})) = -\theta^n$ w $\theta=1, d\theta/d\xi=0$ @ $\xi=0$ (18)

n=0 $\frac{1}{\xi^2} \frac{d}{d\xi} (\xi^2 \frac{d\theta}{d\xi}) = -1$
 $\xi^2 \frac{d\theta}{d\xi} = -\xi^3/3 + C$ $\xrightarrow{0 \text{ from b.c.}} C=0$
 $\theta = -\xi^3/6$

As surface is at $\theta=0$, this occurs at $\xi = \sqrt{6}$

n=1 Rewrite eqn w $\chi = \theta \xi$
 $\therefore d\theta/d\xi = \frac{1}{\xi} d\chi/d\xi - \frac{\chi}{\xi^2}$
 $\therefore \frac{1}{\xi^2} \frac{d}{d\xi} (\xi^2 \frac{d\theta}{d\xi}) = \frac{1}{\xi^2} \frac{d}{d\xi} (\xi \frac{d\chi}{d\xi} - \chi) = \frac{1}{\xi} \frac{d^2\chi}{d\xi^2}$
 $\therefore \frac{d^2\chi}{d\xi^2} = -\xi (\chi/\xi)^n$

For $n=1$ $= -\chi$
 $\therefore \chi = A \sin(\xi + \beta)$ $\xrightarrow{0 \text{ from b.c.}}$
 $\therefore \theta = \frac{A}{\xi} \sin \xi$

So surface is at $\xi = \pi$

n=5 Rewrite eqn w $x = \xi^{-1}$
 $\therefore \frac{1}{\xi^2} \frac{d}{d\xi} (\xi^2 \frac{d\theta}{d\xi}) = x^4 \frac{d^2\theta}{dx^2} = -\theta^n$

Letting $t = \ln x$ and $z = (\frac{x}{2})^{1/2}$
 $\therefore d^2\theta/dx^2 = (\frac{1}{2})^{1/2} [x^{-1/2} d^2z/dx^2 + x^{-3/2} dz/dx - \frac{1}{4} z x^{-3/2}]$
 $= (\frac{1}{2})^{1/2} x^{-3/2} [d^2z/dt^2 - \frac{1}{4} z]$

So the eqn for $n=5$ becomes $d^2z/dt^2 = \frac{1}{4} z(1-z^4)$ [NB x 's cancel]

Multiply by dz/dt $\therefore \frac{d}{dt} [\frac{1}{2} (dz/dt)^2] = \frac{1}{4} z(1-z^4) dz/dt$
 $\therefore \frac{1}{2} (dz/dt)^2 = \frac{1}{8} z^2 - \frac{1}{24} z^6 + C$ $\xrightarrow{0 \text{ from b.c.}}$
 $\int \frac{dz}{z(1-z^4)^{1/2}} = -\frac{1}{2} \int dt$

Letting $\frac{1}{3} z^4 = \sin^2 \phi$ gives $\tan \phi/2 = Ce^{-t}$
 $\therefore \theta = (1 + \frac{1}{3} \xi^2)^{-1/2}$

So surface is at $\xi \rightarrow \infty$

n=∞ $p = K\rho \rightarrow$ isothermal gas sphere, $k = \frac{Rm}{\mu T}$

Need to rederive eq:

Mom eq $\rightarrow d\Phi/dr = -K \frac{1}{\rho} d\rho/dr$ the solⁿ to which is $\ln \rho/\rho_c = \frac{1}{k} [\Phi_c - \Phi]$

Poisson's eq $\rightarrow \frac{1}{r^2} \frac{d}{dr} [r^2 (-k \frac{1}{\rho} \frac{d\rho}{dr})] = 4\pi G \rho$

Letting $\rho = \rho_c e^{-\psi}$ and $r = a \xi$ where $a = [\frac{k}{4\pi G \rho_c}]^{1/2}$

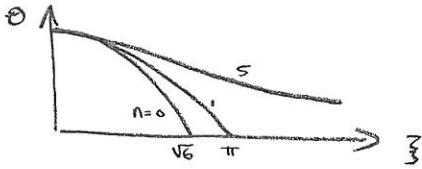
$\rightarrow \frac{1}{\xi^2} \frac{d}{d\xi} (\xi^2 \frac{d\psi}{d\xi}) = e^{-\psi}$ w b.c. $\psi = d\psi/d\xi = 0$ at $\xi = 0$

At large r , solⁿ tends to $\rho \propto r^{-2}$, so mass in isothermal sphere $\rightarrow \infty$ as $r \rightarrow \infty$ and must be truncated at some radius, and to exist in hydrostatic $\hat{=}$ must be embedded in external medium of appropriate pressure

\rightarrow Bonnor-Ebert sphere

NB let $\rho \propto r^{-2}$
 $\rightarrow x = -2$

Scaling relations for polytropes



All stars w same "n" have same $\theta(z)$ profile
 However different stars have different ρ_c or K and so different radii and masses, since these determine how dimensionless variables translate into physical quantities

Remember $\rho = \rho_c \theta^n$
 $r = \left[\frac{(n+1)K}{4\pi G \rho_c^{1-1/n}} \right]^{1/2} \zeta$

Total mass $M = \int_0^{r_{max}} 4\pi r^2 \rho dr$
 $= 4\pi \rho_c \left[\frac{(n+1)K}{4\pi G \rho_c^{1-1/n}} \right]^{3/2} \left(\int_0^{\zeta_{max}} \theta^n \zeta^2 d\zeta \right)$

Radius $R = \left[\frac{(n+1)K}{4\pi G \rho_c^{1-1/n}} \right]^{1/2} \zeta_{max}$ same for stars w same "n"

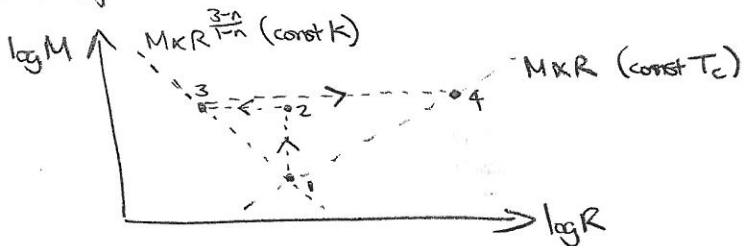
If $K = \text{same}$ for all stars, then $M \propto \rho_c^{\frac{1}{2}(\frac{3}{n}-1)}$, $R \propto \rho_c^{\frac{1}{2}(\frac{1}{n}-1)}$ $\therefore M \propto R^{\frac{3-n}{1-n}}$

Might expect this to be true for fully convective stars for which e.o.s. is nearly adiabatic
 But for monatomic gas $\gamma = 5/3 \rightarrow n = 3/2$ \therefore predicts $M \propto R^{-3}$ ie more massive stars are smaller! whereas $M \propto R$ is observed

If $T_c = \text{same}$ for all stars then:
 as $T_c = \frac{M}{R_*} P_c / \rho_c = \frac{M}{R_*} K \rho_c^{1/n}$
 $\rightarrow K \propto \rho_c^{-1/n}$
 $\therefore M \propto \rho_c^{-1/2}$, $R \propto \rho_c^{-1/2}$ and $M \propto R$

Response of star to mass gain

- Timescale to reach thermal equilibrium is the energy contents of the star divided by its luminosity
 ie, $t_{th} \approx GM^2/RL \rightarrow 30 \text{ Myr}$
- Timescale to reach hydrostatic equilibrium is the time for sound wave to cross the star
 ie, $t_h \approx R/c_s \approx 1 \text{ day}$
- So response of star to mass gain is to evolve at const K on timescale t_h
 Then readjust "K" on timescale t_{th} so as to achieve thermal \Rightarrow



- 1 = initial state ($t=0$)
- 2 = state after perturbation ($t=0$)
- 3 = intermediate state ($t=t_h$)
- 4 = final state ($t=t_{th}$)

eg, for $n = 3/2$, star shrinks then expands.

Eq 1

Consider star rotating at angular velocity Ω
Drop non-rotating mass ΔM onto star - what is new ang vel $\Omega + \Delta\Omega$?

Angular momentum $J \propto MR^2\Omega$ is conserved

If $M \propto R^{3-n}$ then $J \propto M^{1+2(1-n)/(3-n)} \Omega$

$$\propto M^{\frac{5-3n}{3-n}} \Omega$$

$$\therefore \Delta J \propto \left(\frac{5-3n}{3-n}\right) M^{\frac{2-2n}{3-n}} \Omega \Delta M + M^{\frac{5-3n}{3-n}} \Delta\Omega = 0$$

$$\therefore \Delta\Omega / \Delta M = -(\Omega/M) \left[\frac{5-3n}{3-n}\right]$$

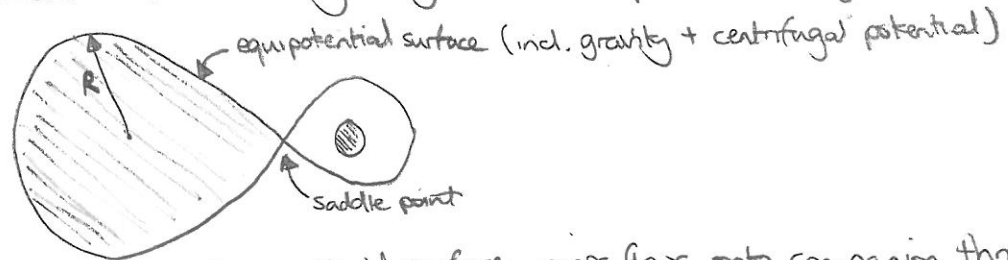
So stars spins up if $3n > 5/3$ and spins down if $n < 5/3$

(Physically larger $n \rightarrow$ squishier e.o.s. \rightarrow star shrinks more \rightarrow spins up to conserve J)

NB on longer timescales star spins down as $M \propto R \rightarrow \Delta\Omega / \Delta M = -3\Omega/M$

Eq 2

Consider star in binary losing mass to companion through Roche Lobe overflow



If star is larger than critical surface, mass flows onto companion through saddle point
eg. late in life as star expands as red giant, or if binary orbit shrinks

Mass loss affects stellar radius s.t. if $M \propto R^{\frac{3-n}{1-n}}$

$$\Delta R / \Delta M \propto \frac{1-n}{3-n} M^{-2/(3-n)} - \text{unimportant}$$

So, if $1 < n < 3$ the star expands as it loses mass

Moreover mass loss means critical surface gets closer to star

\rightarrow runaway process

Sound waves

(21)

Principal mechanism by which disturbances propagate in fluids

[eg. talking, shock waves, mass added to star - exception is Alfvén waves in magnetised media]

- ③ • Consider uniform medium w no external forces

$$\partial \rho / \partial t + \nabla \cdot (\rho \underline{u}) = 0$$

$$\partial \underline{u} / \partial t + \underline{u} \cdot \nabla \underline{u} = -\frac{1}{\rho} \nabla p$$

- ④ • Unperturbed state of fluid in equilibrium ($\partial/\partial t = 0$) in $p = p_0$, $\rho = \rho_0$, $\underline{u} = 0$

- ⑤ • Consider a Lagrangian perturbation to this equilibrium

$$p = p_0 + \Delta p, \quad \rho = \rho_0 + \Delta \rho, \quad \underline{u} = \Delta \underline{u}$$

(ie. a perturbation to fluid elements)

- The Eulerian pertⁿ, which is what should be substituted into above eqns, is

$$\delta X = \Delta X - \underline{\xi} \cdot \nabla X = \Delta X - \underline{\xi} \cdot \nabla X_0$$

ie. the quantity X at fixed point P changes both because pertⁿ has changed local fluid element and because the pertⁿ may have moved a fluid element to different unperturbed value of X to point P

$$\underline{\xi} = \text{small displacement of fluid element at } P \text{ due to pert}^n$$

[NB derivation is analogous to that for D/Dt]

- Here, as unperturbed quantities are uniform ($\nabla X_0 = 0$) $\Rightarrow \delta X = \Delta X$

- ① • So to 1st order:

$$\begin{aligned} \partial \Delta \rho / \partial t + \rho_0 \nabla \cdot \Delta \underline{u} &= 0 \\ \partial \Delta \underline{u} / \partial t &= -\frac{1}{\rho_0} \nabla \Delta p \end{aligned}$$

If fluid is barotropic $\Delta p = \frac{dp}{d\rho} \Delta \rho$

$$\partial \Delta \underline{u} / \partial t = -\left(\frac{dp}{d\rho}\right) \frac{1}{\rho_0} \nabla \Delta \rho$$

- ② $\partial/\partial t \Rightarrow \partial^2 \Delta \rho / \partial t^2 + \rho_0 \nabla \cdot \partial \Delta \underline{u} / \partial t = 0$

$$\therefore \frac{\partial^2 \Delta \rho / \partial t^2}{\left(\frac{dp}{d\rho}\right)} = \nabla^2 \Delta \rho \quad \text{which is a wave equation}$$

- ③ • Solution: $\Delta \rho = \Delta \rho_0 e^{i(kx - \omega t)}$ (ie. in 1D, $kx \rightarrow \underline{k} \cdot \underline{r}$ in higher dimensions)

Substituting into wave eqⁿ $\rightarrow \omega^2/k^2 = dp/d\rho$

As points of constant phase ($kx - \omega t = \text{const}$) propagate at a speed ω/k

the wave travels at speed $c_s = \omega/k$

$$= \sqrt{dp/d\rho} = \text{Sound speed}$$

$$k = \text{wavenumber} = 2\pi/\lambda$$

$$\omega = \text{angular frequency} = 2\pi\nu$$

• All parts obey same wave eqn = eg. substituting soln into mom. eq.

$$\rightarrow \Delta u = \left(\frac{\Delta p_0}{\rho_0}\right) \frac{1}{k} e^{i(kx - \omega t)} = \left(\frac{\Delta p_0}{\rho_0}\right) c_s$$

As $\Delta p/\rho_0 \ll 1$ the disturbance propagates faster than the speed of individual fluid elements
[eg. disturbance produced by my voice does not make air move at 600 mph!]

• Sound waves are longitudinal waves that propagate because a density pertⁿ → pressure gradient → accelerⁿ of fluid elements → fluid velocities that induce density pertⁿ [longitudinal because pressure forces act \perp to surfaces in fluid]

• Stiff e.o.s. (high $dp/d\rho$) → large restoring force → rapid propagation

Isothermal vs adiabatic

• $c_s^2 = dp/d\rho|_T$ occurs if density pertⁿ pass heat to each other ^{eg by conduction or radⁿ} faster than $t_{oscill} = \text{timescale } \frac{1}{\omega}$ for fluid elements
 $= \left(\frac{R_0}{\mu}\right) T$

• $c_s^2 = dp/d\rho|_s$ if no heat exchange between fluid elements (ie heat transfer timescale $\gg \frac{1}{\omega}$) s.t. compressions heat up by pdV work.
 $= \gamma \left(\frac{R_0}{\mu}\right) T$

• thermal behaviour of sound waves not necessarily same as unperturbed medium
eg sound waves in air behave adiabatically in isothermal atmosphere.

NB in this example sound waves are non-dispersive → c_s is not a function of ω
ie all frequencies propagate at same rate.

[the shape of a packet of waves of diff. freqs. is preserved as it propagates]



Stratified atmosphere

Same eqs as before but include const. gravity in z dir
 x and y components of eqns unaffected, so horizontal sound waves unaffected

a) $\rightarrow \partial p / \partial t + \partial / \partial z (\rho u) = 0$
 $\partial u / \partial t + u \partial u / \partial z = -\frac{1}{\rho} \partial p / \partial z - g$

b) Equilibrium sol, assuming isothermal $p = A\rho$ where $A = R_0 T / M$

$u_0 = 0$
 $\rho_0(z) = \tilde{\rho} e^{-z/H}$ where $H = A/g$

c) Perturb with Lagrangian pert, but need to 1/P Eulerian pert into eqns

$\rho = \rho_0 + \delta\rho = \rho_0 + \Delta\rho - \tilde{z} \cdot \nabla \rho_0$ [where \tilde{z} is Lagrangian displacement of fluid element due to pert]
 $= \rho_0 + \Delta\rho - \tilde{z}_z \partial \rho_0 / \partial z$ [since only considering z varies here]
 $= \rho_0 (1 - \tilde{z}_z / H + \Delta\rho / \rho_0)$
 $u = u_0 + \delta u = \Delta u - \tilde{z} \cdot \nabla u$ ← Similarly $p = p_0 + \Delta p - \tilde{z}_z \partial p_0 / \partial z$
 $= \Delta u_z - \tilde{z}_z \partial \Delta u_z / \partial z$ $= A\rho_0 + \Delta p - A \tilde{z}_z \partial \rho_0 / \partial z$
 $= \Delta u_z$ $= A\rho_0 (1 - \tilde{z}_z / H) + \Delta p$

d) Continuity $\rightarrow \partial [\rho_0 + \Delta\rho - \tilde{z}_z \partial \rho_0 / \partial z] / \partial t + \partial / \partial z [(p_0 + \Delta p - \tilde{z}_z \partial p_0 / \partial z) \Delta u_z] = 0$
 $\therefore \partial \Delta\rho / \partial t - (\partial \tilde{z}_z / \partial t) (\partial \rho_0 / \partial z) + \Delta u_z \partial \rho_0 / \partial z + \rho_0 \partial \Delta u_z / \partial z = 0$

But $\Delta u = D\tilde{z} / Dt = \partial \tilde{z} / \partial t + u \cdot \nabla \tilde{z}$
 $\approx \partial \tilde{z} / \partial t$ [to first order as $u_0 = 0$]

This means above terms cancel

$\therefore \partial \Delta\rho / \partial t + \rho_0 \partial \Delta u_z / \partial z = 0$

Momentum $\Rightarrow \partial \Delta u_z / \partial t = -\rho_0^{-1} [1 - \tilde{z}_z / H + \Delta\rho / \rho_0] \frac{\partial}{\partial z} [A\rho_0 (1 - \tilde{z}_z / H) + \Delta p] - g$
 $= -\rho_0^{-1} [1 + \tilde{z}_z / H + \Delta\rho / \rho_0] [-\frac{A}{H} \rho_0 + \frac{\partial \Delta p}{\partial z} + \frac{\tilde{z}_z A \rho_0}{H} - \frac{A \rho_0}{H} \frac{\partial \tilde{z}_z}{\partial z}] - g$
 $= -\rho_0^{-1} [\frac{\partial \Delta p}{\partial z} + (\frac{A \rho_0}{H}) (-1 + \frac{\tilde{z}_z}{H} - \frac{\partial \tilde{z}_z}{\partial z} - \frac{\tilde{z}_z}{H} - \frac{\Delta\rho}{\rho_0})] - g$

① cancels because this is \approx sol = (when pert is zero)

② cancels straight forwardly

③ cancels from continuity as $\partial \Delta\rho / \partial t + \rho_0 \partial (\partial \tilde{z}_z / \partial t) / \partial z = 0$

$\Delta p / \rho_0 = -\partial \tilde{z}_z / \partial z$ (multiplying through by ρ_0)

$\therefore \partial \Delta u_z / \partial t = -\rho_0^{-1} \partial \Delta p / \partial z = -\frac{c_s^2}{\rho_0} \partial \Delta p / \partial z$ where $c_s^2 = \frac{dp}{d\rho}$ fluid elements obey eq of state

e) Wave eqn $\partial / \partial t$ (cont) is $\partial^2 \Delta p / \partial t^2 + \rho_0 \partial^2 \rho u_z / \partial z^2 = 0$
 $\therefore \partial^2 \Delta p / \partial t^2 = c_s^2 \partial^2 \Delta p / \partial z^2 + \frac{c_s^2}{\rho_0} (\partial \rho_0 / \partial z) (\partial \Delta p / \partial z) = 0$
 $\therefore \partial^2 \Delta p / \partial t^2 - c_s^2 \partial^2 \Delta p / \partial z^2 - \frac{c_s^2}{H} \partial \Delta p / \partial z = 0$

f) Dispersion relation

Let $\Delta p \propto e^{i(kz - \omega t)}$

then $-\omega^2 + c_u^2 k^2 - c_u^2 ik/H = 0$ is the dispersion relⁿ betw angular freq. ω and wave number k

$\therefore k^2 - \frac{i}{H}k - \frac{\omega^2}{c_u^2} = 0$ i.e. quadratic in k

$\therefore k = \frac{i}{2H} \pm \sqrt{\frac{\omega^2}{c_u^2} - \frac{1}{4H^2}}$

For $\omega > c_u/2H$

$\text{Im}(k) = \frac{1}{2H}$

$\text{Re}(k) = \pm \sqrt{\frac{\omega^2}{c_u^2} - \frac{1}{4H^2}} = \pm k_r$

$\therefore \Delta p \propto e^{-z/2H} e^{i[K_r z - \omega t]}$

Although this is exponentially decaying w/ z , note that $p_0 \propto e^{-z/H}$ so fractional varⁿ $\Delta p/p_0$ grows w/ height until $\Delta p/p_0 \rightarrow 1$, linear treatment breaks down and shock forms (or more realistically energy is dissipated and wave is damped)

This looks like standard wave eqn, but k_r is a fⁿ of ω , so lines of const phase propagate at speed $V_{ph} = \frac{\omega}{k_r} = c_u [1 + (\frac{1}{2k_r H})^2]^{-1/2}$

\therefore different frequencies travel at diff speeds \rightarrow dispersive

eg. consider two frequencies $\omega + \delta\omega$ w/ corresponding $k_r + \delta k_r$
and $\omega - \delta\omega$ $k_r - \delta k_r$

the superposed amplitude is $\sin[k_r z - \omega t + (\delta k_r z - \delta\omega t)] + \sin[k_r z - \omega t - (\delta k_r z - \delta\omega t)]$
 $= 2 \sin(k_r z - \omega t) \cos(\delta k_r z - \delta\omega t)$

wave propagating at V_{ph} modulation of wave amplitude propagating at $\delta\omega/\delta k_r =$ group velocity

$V_{gr} = \frac{d\omega}{dk_r} = c_u [1 + (\frac{1}{2k_r H})^2]^{-1/2}$ NB $V_{gr} = V_A$ in non-dispersive medium ($H \rightarrow \infty$)

For $\omega < c_u/2H$

$\text{Re}(k) = 0$ and $\text{Im}(k) > 0$

$\therefore \Delta p \propto e^{-k_i z} e^{i\omega t}$

\rightarrow all points oscillate in phase with an amplitude that depends on z
= standing wave

NB wavelength of travelling wave is same order as H , so change in properties of atmosphere over wavelength are significant.

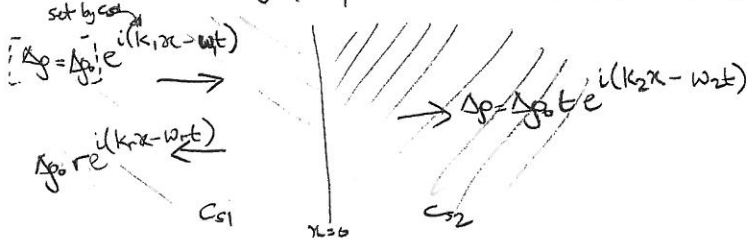
see handout for general approach to wave propagation problems.

Transmission at interfaces

[previously considered $c_s = \text{const}$]

(25)

Consider non dispersive medium w boundary at $x=0$ and sound wave travelling L to R of ang. freq. ω . Some fraction is transmitted, some reflected:



At boundary ($x=0$), accelerations are finite, so $\omega_2 = \omega_r = \omega_1 = \omega$ [NB $-\omega_1^2 e^{-i\omega_1 t} - \omega_r^2 e^{-i\omega_r t} = -\omega_1^2 e^{-i\omega t}$]

Reflected wave travels at c_{s1} in negative x dir, so $k_r = -k_1$

Amplitude at ($x=0$) is continuous, so $1 + r = t$

Derivative at ($x=0$) is continuous, so $k_1(1-r) = k_2 t$

$$t = 2k_1 / (k_1 + k_2)$$

$$r = (k_1 - k_2) / (k_1 + k_2)$$

$$\text{NB } k_i = \omega / c_{si}$$

- So if $c_{s2} > c_{s1}$, $k_2 < k_1$, so $r > 0$ and reflected wave is in phase w incident wave
 $c_{s2} \ll c_{s1}$, $k_2 \gg k_1$, so $t \rightarrow 0$ (\therefore difficult to excite disturbances in cold GMC through sand waves)

• Kinetic energy flux in wave is $\propto \rho (\delta u)^2 c_s$
 Approximate $\sim \frac{\rho}{c_s} (\delta u)^2$

Since ρ is constant across boundary $KE_i \propto \frac{1}{c_{s1}} \times I^2 \propto k_1$

$$KE_t \propto \frac{1}{c_{s2}} E^2 \propto k_2 (2k_1)^2 / (k_1 + k_2)$$

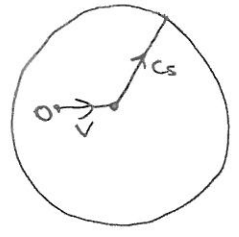
$$KE_r \propto \frac{1}{c_{s1}} r^2 \propto k_1 (k_1 - k_2)^2 / (k_1 + k_2)$$

$$\rightarrow KE_t + KE_r = KE_i$$

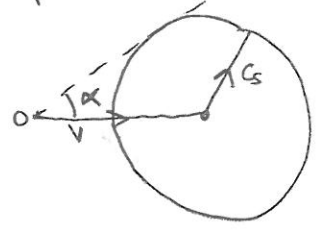
Shocks

Disturbances propagate at speed c_s relative to fluid (as per wave Lagrangian)
 Consider observer at source of spherical disturbance also fluid flowing at v
 The velocity of disturbance relative to observer:

Subsonic



Supersonic

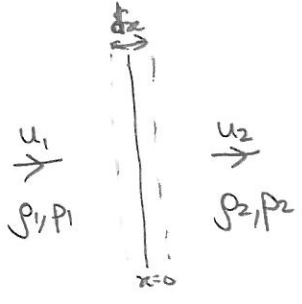


Disturbance can propagate 4π sr if subsonic
 into Mach cone defined by α if supersonic

∴ disturbances cannot propagate upstream from obstacle in supersonic flow, so flow is undisturbed until it reaches obstacle where properties change discontinuously in shock

Rankine-Hugoniot relations

Consider shock at $x=0$, width δx , normal in x -dir



PH1 Continuity

$$\frac{\partial \rho}{\partial t} + \frac{\partial}{\partial x}(\rho u_x) = 0$$

Integrate over layer of thickness δx

$$\frac{\partial}{\partial t} \int \rho dx + (\rho u_x)_{x+\delta x} - (\rho u_x)_{x-\delta x} = 0$$

As mass doesn't accumulate $\rho_1 u_1 = \rho_2 u_2$

PH2 Momentum

$$\frac{\partial}{\partial t}(\rho u_x) = -\frac{\partial}{\partial x}(\rho u_x^2 + p) - \rho \frac{\partial \Phi}{\partial x}$$

Integrate over δx , noting that Φ is continuous and that continuity means LHS = 0

$$p_1 + \rho_1 u_1^2 = p_2 + \rho_2 u_2^2$$

ie sum of ram and thermal pressures is constant

NB PH3 if there had been an additional u_y, u_z component $\rho_1 u_1 u_y = \rho_2 u_1 u_y \rightarrow u_y = u_y$ (and likewise for z)
 so easiest to do problem (as above) in frame moving at u_y, u_z

PH3 Energy

If shock is adiabatic, so $Q_{\text{rad}} = 0$

$$\frac{\partial E}{\partial t} + \frac{\partial}{\partial x}[(E+p)u_x] = \rho \frac{\partial \Phi}{\partial t}$$

Integrate over δx , noting Energy doesn't accumulate

$$(E_1 + p_1)u_1 = (E_2 + p_2)u_2$$

As $E = \rho(\frac{1}{2}u^2 + \epsilon + \Phi)$ (then is continuity, noting Φ is continuous)

$$\frac{1}{2}u_1^2 + \epsilon_1 + p_1/\rho_1 = \frac{1}{2}u_2^2 + \epsilon_2 + p_2/\rho_2$$

∴ showing how KE can be converted into enthalpy ϵ
 $\epsilon = \epsilon + p/\rho$

For ideal gas $E = CvT = \frac{Cv}{(\gamma-1)} P/\rho = \frac{1}{\gamma-1} P/\rho$

So for an adiabatic shock: $\frac{1}{2} u_1^2 + \left(\frac{\gamma}{\gamma-1}\right) P_1/\rho_1 = \frac{1}{2} u_2^2 + \left(\frac{\gamma}{\gamma-1}\right) P_2/\rho_2$

Combine to get ρ_2/ρ_1

If (R1) $\rho_1 u_1 = j$ then mom (R2) is $P_1 + j^2/\rho_1 = P_2 + j^2/\rho_2$
 $\therefore j^2 = (P_2 - P_1) \left(\frac{1}{\rho_1} - \frac{1}{\rho_2}\right)^{-1}$
 energy (R3) is $\frac{1}{2} j^2/\rho_1^2 + \left(\frac{\gamma}{\gamma-1}\right) \frac{P_1}{\rho_1} = \frac{1}{2} j^2/\rho_2^2 + \left(\frac{\gamma}{\gamma-1}\right) \frac{P_2}{\rho_2}$
 $\therefore \frac{1}{2} (P_2 - P_1) \left(\frac{1}{\rho_1} - \frac{1}{\rho_2}\right)^{-1} \left(\frac{1}{\rho_1^2} - \frac{1}{\rho_2^2}\right) = \left(\frac{\gamma}{\gamma-1}\right) \left(\frac{P_2}{\rho_2} - \frac{P_1}{\rho_1}\right)$
 $\therefore \frac{1}{\rho_2} \left[P_2 \left(\frac{1}{2} - \frac{\gamma}{\gamma-1}\right) - \frac{1}{2} P_1 \right] = \frac{1}{\rho_1} \left[P_1 \left(\frac{1}{2} - \frac{\gamma}{\gamma-1}\right) - \frac{1}{2} P_2 \right]$
 $\therefore \rho_2/\rho_1 = u_1/u_2 = \frac{(\gamma+1)P_2 + (\gamma-1)P_1}{(\gamma+1)P_1 + (\gamma-1)P_2}$

So in limit for strong shocks for which $P_2 \gg P_1$ (neglect upstream pressure)
 $\rightarrow \rho_2/\rho_1 = \frac{\gamma+1}{\gamma-1} = 4$ for $\gamma = 5/3$

This is the maximum possible density contrast (as a larger thermal pressure behind the shock prevents it from being compressed too much).

Rewrite in terms of Mach # $M = u/c_s$

As adiabatic $c_s^2 = \gamma P/\rho$

(R1) $\Rightarrow \rho_1 M_1 c_{s1} = \rho_2 M_2 c_{s2}$
 (R2) $\Rightarrow \rho_1 c_{s1}^2/\gamma + \rho_1 M_1^2 c_{s1}^2 = \rho_2 c_{s2}^2/\gamma + \rho_2 M_2^2 c_{s2}^2$ [just substituting for p and u!]
 $\therefore M_1^2 c_{s1} \left[M_1^2 + \frac{1}{\gamma} \right] = M_2^2 c_{s2} \left[M_2^2 + \frac{1}{\gamma} \right]$ [eliminate p by dividing by (R1) (LHS by LHS, RHS by RHS)]
 (R3) $\Rightarrow \frac{1}{2} M_1^2 c_{s1}^2 + \left(\frac{\gamma}{\gamma-1}\right) c_{s1}^2 = \frac{1}{2} M_2^2 c_{s2}^2 + \left(\frac{\gamma}{\gamma-1}\right) c_{s2}^2$
 $\therefore \left(M_1^2 + \frac{2}{\gamma-1}\right) M_1^2 \left(M_1^2 + \frac{1}{\gamma}\right)^2 = \left(M_2^2 + \frac{2}{\gamma-1}\right) M_2^2 \left(M_2^2 + \frac{1}{\gamma}\right)^2$ [eliminate c_s by dividing by (R2)^2]

Expanding out:

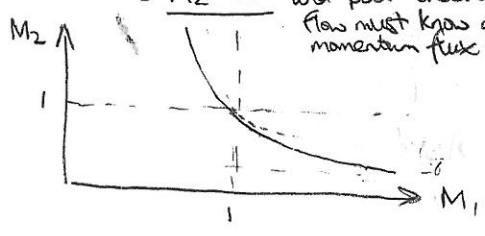
$(M_1^2 - M_2^2) \left[2M_1^2 M_2^2 \left(\frac{1}{\gamma-1} - \frac{1}{\gamma}\right) - \frac{1}{\gamma} (M_1^2 + M_2^2) - \frac{2}{\gamma^2 (\gamma-1)} \right] = 0$ [after some effort]

Thus either $M_1 = M_2$ and there's no shock
 or $M_2^2 = \frac{2 + (\gamma-1)M_1^2}{2\gamma M_1^2 - (\gamma-1)}$

Rewrite, letting $B = 2\gamma M_1^2 - (\gamma-1) = 1 + \gamma(2M_1^2 - 1)$
 $\rightarrow M_2^2 = \frac{1}{B} [B + (1+\gamma)(1-M_1^2)]$

So, if $M_1 > 1$ (and $\gamma > 1$)
 $\rightarrow M_2 < 1$

and post shock flow is subsonic, which is expected as post shock flow must know about conditions at the shock to set conditions like mass and momentum flux rates.



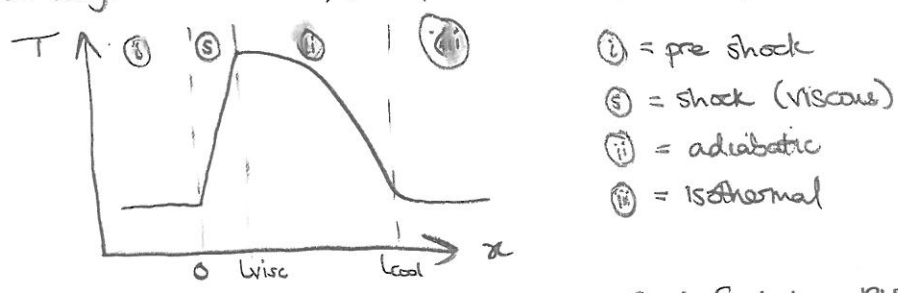
As $M_1 \rightarrow \infty$, $M_2 \rightarrow \frac{\gamma-1}{2\gamma}$

The shock itself

- The RH cond's only specify cond's on either side of shock where inviscid, equs are valid
- Within the shock it must be a different story as entropy of post-shock fluid has changed
 - $dS = C_v d \ln(p/\rho^\gamma) \neq 0$ [not specified as such, just that $p = K\rho^\gamma$ outside shock, not what K is]
- As entropy can only increase application of the RH rel's shows that the supersonic flow must be upstream of shock (ie. $M_1 > 1$)
 - ie shocks convert cold fast fluid into hot slow fluid, not the other way around
- How? by an irreversible process such as viscosity converting KE into heat, and shock thickness is set by length over which viscosity converts mechanical energy into heat, L_{visc}

Isothermal shocks

- Adiabatic assumption may be ok just after shock, but $\dot{Q} \neq 0$ and shocked gas may cool over some length scale L_{cool} , perhaps back to original temperature



- (i) = pre shock
- (s) = shock (viscous)
- (ii) = adiabatic
- (iii) = isothermal

- To work out cond's in isothermal region note that first two RH rel's still hold

(1) $\rho_1 u_1 = \rho_3 u_3$

(2) $p_1 + \rho_1 u_1^2 = p_3 + \rho_3 u_3^2$

- Third doesn't as gas loses heat, but $T_3 = T_1$
- $c_{s3} = c_{s1} = (p_1/\rho_1)^{1/2}$ NB assuming pre-shock gas also behaves isothermally

So $\rho_1 c_{s1}^2 + \rho_1 u_1^2 = \rho_3 c_{s3}^2 + \rho_3 u_3^2$ [putting p into (2)]

$(c_{s1}^2 + \rho_1 u_1^2) u_3 = (c_{s1}^2 + \rho_3 u_3^2) u_1$ [dividing by (1) LHS by LHS etc]

$(u_3 - u_1)(c_{s1}^2 - u_1 u_3) = 0$

- Thus either $u_3 = u_1$ ie no shock
- or $u_1 u_3 = c_{s1}^2$
- $\rho_3/\rho_1 = u_1/u_3 = (u_1/c_{s1})^2 = M_1^2$

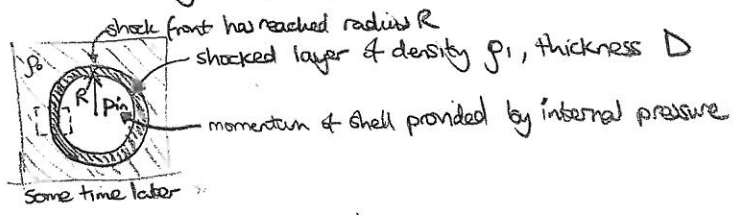
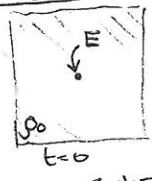
So density ratio can be high in high Mach # isothermal shock and $M_3 < M_1$ as before

* NB $dQ = TdS = C_v dT - \frac{p}{\rho^2} d\rho$
 $dS = C_v d \ln T - \frac{p}{\rho} d \ln \rho$
 $= C_v d \ln T / (d \ln T / d \ln \rho) / C_v$
 $= C_v d \ln \frac{p}{\rho^\gamma}$
 $= C_v d \ln p / \rho^\gamma$

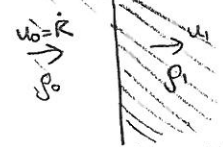
Blast waves

eg. SN explodes depositing energy E in ISM of uniform density ρ_0
 shocked medium expands sweeping up gas \rightarrow ISM has swiss-cheese-like structure

Simple model



In frame moving w/ shock



NS ($\dot{R} \leftarrow$) in initial frame

- Ignore thermal pressure from medium (ie strong shock, $\rho_1 \gg \rho_0$),
 \rightarrow if adiabatic $\rho_1/\rho_0 = \frac{\gamma+1}{\gamma-1}$
- Mass conservation $\rho_1 u_1 / \rho_0 u_0 = \rho_0 / \rho_1 = \frac{\gamma-1}{\gamma+1}$
 \rightarrow velocity of shocked layer in initial frame $U = \dot{R} - u_1 = u_0 - u_1 = \left(\frac{2}{\gamma+1}\right)\dot{R}$
- Mass in shocked layer is sum of swept up medium $\frac{4\pi}{3}\rho_0 R^3$
 NB if D/R is small this is $4\pi R^2 D \rho_1 \rightarrow D/R = \frac{1}{3} \left(\frac{\gamma-1}{\gamma+1}\right) \ll 1$ justifying assumption of thin shell
- Rate of change of momentum of shocked layer is $\frac{d}{dt} \left[\frac{4\pi}{3} \rho_0 R^3 \left(\frac{2}{\gamma+1}\right) \dot{R} \right]$
- Assume provided by internal pressure that is some fraction α of pressure in shocked layer

where $P_{in} = \alpha P_1$
 (R12) $P_1 + \rho_1 u_1^2 = \rho_0^2 + \rho_0 u_0^2$ (assumed negligible)
 $\therefore P_1 = \rho_0 \dot{R}^2 \left[1 - \left(\frac{\gamma-1}{\gamma+1}\right) \right] = \left(\frac{2}{\gamma+1}\right) \rho_0 u_0^2$

Thus $\left(\frac{4\pi}{3}\right) \left(\frac{2}{\gamma+1}\right) \rho_0 \frac{d}{dt} (R^3 \dot{R}) = 4\pi R^2 \alpha \left(\frac{2}{\gamma+1}\right) \rho_0 \dot{R}^2$ [ie pressure x area]

$\frac{d}{dt} (R^3 \dot{R}) = 3\alpha R^2 \dot{R}^2$
 Let $R \propto t^b \rightarrow b(4b-1)t^{4b-2} = 3\alpha b^2 t^{4b-2}$
 $\therefore b = \frac{1}{4-3\alpha}$

$\therefore R \propto t^{\frac{1}{4-3\alpha}}$

- If blast wave is adiabatic, energy E is conserved and must be taken up in
 - KE of shell = $\frac{1}{2} \left(\frac{4\pi}{3}\rho_0 R^3\right) U^2$
 - Internal energy = As internal energy (just vol) is $\frac{P}{\gamma-1}$ (NB our simple assumptions are ρ are const over vol!)
 $= \frac{4\pi}{3} R^3 \frac{P_{in}}{\gamma-1} + 4\pi R^2 D \frac{P_1}{\gamma-1}$
 $= \frac{4\pi}{3} R^3 P_1 \left[\frac{\alpha}{\gamma-1} + \frac{1}{\gamma+1} \right]$
- $E = \frac{4\pi}{3} R^3 \left[\frac{1}{2} \rho_0 \left(\frac{2}{\gamma+1}\right)^2 \dot{R}^2 + \left[\frac{1}{\gamma+1} + \frac{\alpha}{\gamma-1} \right] \left(\frac{2}{\gamma+1}\right) \rho_0 \dot{R}^2 \right]$
 $\propto R^3 \dot{R}^2$
 $\propto t^{\frac{6\alpha-3}{4-3\alpha}}$
 So for $\dot{E}=0$, $\alpha = 1/2$ and $R \propto t^{2/5}$

Dimensional arguments

Problem defined by E ($ML^2 T^{-2}$)
 ρ_0 (ML^{-3})

So no natural length scale, but at time t can define a length dimension $\lambda = \left(\frac{Et^2}{\rho_0}\right)^{1/5}$,
 and can assume $sd = is$ rbf - similar - s.t. E, ρ_0

$X(r, t) = X_1(r, t, \text{dimensional values of problem}) X'(\xi)$ where $\xi = r/\lambda$ and $X_1 = K \times$ combination that gives appropriate dimensions
 (ie dist of X at given time is a scaled version of that at other times)

Thus shock front will occur at some constant $\xi_0 \therefore R = \xi_0 \lambda = \xi_0 \left(\frac{E}{\rho_0}\right)^{1/5} t^{2/5}$

eg) if $\xi_0 \approx 1$ then an SN ejecting $1M_\odot$ at 10^4 km/s ($E = 10^{44}$ J) into ISM of $\rho_0 \approx 10^{-21}$ kg/m³ $\rightarrow R = 10^{13}$ t^{2/5}
 or in astronomical units $R \approx 0.3 t^{2/5}$ pc (t in yr), $\dot{R} \approx 10^5 t^{-3/5}$ km/s

NB, sol = only valid for $t > 100$ yr (for $\dot{R} < 10^4$ km/s and for swept up mass $>$ SN ejecta)
 $t < 10^5$ yr (after which energy losses important)

Sedov Similarity Solution

Define dimensionless variables inside the shock scaled s.t. $\rho'(\xi_0), p'(\xi_0) = u'(\xi_0) = 1$

Thus $\rho(r, t) = K_1 \rho_0'(\xi)$ [as physical quantities inside shock only depend on ξ]
 From (RH1) $K_1 = \left(\frac{\gamma+1}{\gamma-1}\right)$

The velocity in inertial frame requires additional (r/t) factor (being only combin of fundamental quantities giving units of velocity)

$u(r, t) = K_2 (r/t) u'(\xi)$

As $\dot{R} = \frac{2}{5} R/t$ and $u(R, t) = \left(\frac{2}{\gamma+1}\right) \dot{R} \rightarrow K_2 = \frac{4}{5(\gamma+1)}$

The pressure requires $\rho(r/t)^2$ factor as it has units of velocity squared

$p(r, t) = K_3 \rho_0 (r/t)^2 p'(\xi)$

As (RH2) gave $p(R, t) = p_1 = \left(\frac{2}{\gamma+1}\right) \rho_0 u_0^2 = \left(\frac{2}{\gamma+1}\right) \rho_0 (\dot{R})^2 \rightarrow K_3 = \frac{8}{25(\gamma+1)}$

*Continuity eqn (sph. symm.) $\frac{\partial \rho}{\partial t} + \frac{1}{r^2} \frac{\partial}{\partial r} (r^2 \rho u) = \frac{\partial \rho}{\partial t} + 2\rho u/r + \rho u \frac{\partial}{\partial r} = 0$

But $\frac{\partial}{\partial t} \Big|_r = \left(\frac{\partial \xi}{\partial t}\right) \frac{\partial}{\partial \xi} \Big|_r = -\frac{2}{5} \frac{\xi}{t} \frac{\partial}{\partial \xi} \Big|_r$ ← NB $\xi = \rho^{1/5} E^{2/5} r^{-2/5}$
 $\frac{\partial}{\partial r} \Big|_t = \left(\frac{\partial \xi}{\partial r}\right) \frac{\partial}{\partial \xi} \Big|_t = \left(\frac{3}{r}\right) \frac{\partial}{\partial \xi} \Big|_t$

So ① is $-\frac{2}{5} \frac{\xi}{t} K_1 \rho_0' \frac{\partial}{\partial \xi}$

② is $2K_1 \rho_0' \frac{1}{r} \rho' u'$

③ requires some care as need to do differential at constant t , so substitute $r = \xi \left(\frac{E}{\rho_0}\right)^{1/5} t^{2/5}$ to get

$\frac{3}{r} \frac{\partial}{\partial \xi} \left[K_1 \rho_0' K_2 \xi \left(\frac{E}{\rho_0}\right)^{1/5} t^{-2/5} u' \right] = \frac{K_1 K_2 \rho_0'}{t} \frac{\partial}{\partial \xi} (\rho' u' \xi) = \frac{K_1 K_2 \rho_0'}{t} \left[\xi \frac{\partial}{\partial \xi} (\rho' u') + \rho' u' \right]$

Combining to $\xi \frac{\partial}{\partial \xi} (\rho' u') - \left(\frac{2}{\gamma+1}\right) [\xi \rho' u' + \xi \frac{\partial}{\partial \xi} (\rho' u')] = 0$

Note that the "r" and "t" cancelled leaving this only in terms of ξ , so similarity solⁿ is poss.

*Momentum $\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial r} = -\frac{1}{\rho} \frac{\partial p}{\partial r}$

$\rightarrow -u' - \frac{2}{5} \frac{\xi}{t} \frac{du'}{d\xi} + \frac{4}{5(\gamma+1)} (u'^2 + u' \xi \frac{du'}{d\xi}) = -\frac{2}{5} \frac{\gamma-1}{\gamma+1} \frac{1}{\rho'} (2\rho' + \xi \frac{d\rho'}{d\xi})$

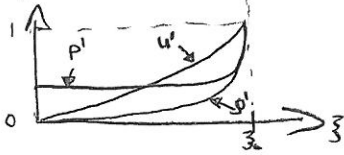
*Energy $D/Dt (p/\rho^\gamma) = 0 \rightarrow \xi \frac{d}{d\xi} (\ln p'/\rho') = \frac{5(\gamma+1)-4u'}{2u'-(\gamma+1)}$

*Boundary conditions $p'(\xi_0) = \rho'(\xi_0) = u'(\xi_0) = 1$

and to find ξ_0 , we $E = \int_0^R \left(\frac{1}{2} \rho u^2 + \frac{p}{\gamma-1}\right) 4\pi r^2 dr$

or in dimensionless variables $\frac{32\pi}{25(\gamma-1)} \int_0^{\xi_0} (\rho' + \rho' u'^2) \xi^4 d\xi = 1$

Solution for $\gamma = 7/5$



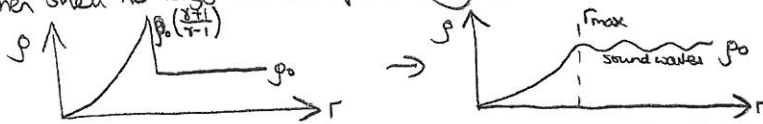
Thus our simple model of most mass in shell, $p_{in}/p_{out} \neq \text{const}$, $u_1 \neq \text{const}$ was ok.

Maximum radius

This solⁿ only applicable if no counter pressure outside shock (st. post shock state is scale free) \rightarrow breaks down when

$$p_1 = \left(\frac{2}{\gamma+1}\right) \rho_0 \dot{R}^2 \approx p_0$$

As $c_{s0}^2 = \gamma p_0 / \rho_0$ this is when $\dot{R} / c_{s0} = \sqrt{\frac{\gamma+1}{2\gamma}}$
 ie when shell no longer moves supersonically wrt medium at which pt sound waves propagate ahead of blast wave



To estimate r_{max} , revisit simple model w/ $\alpha = 1/2 \rightarrow E = \frac{4\pi}{3} R^3 \rho_0 \dot{R}^2 \frac{5\gamma-3}{(\gamma^2-1)(\gamma+1)}$

So at point where $p_1 \approx p_0$

$$= \left[\frac{4\pi}{3} R_{max}^3 \rho_0 \frac{c_{s0}^2}{\gamma(\gamma-1)} \right] \frac{5\gamma-3}{2(\gamma+1)}$$

As $E_0 = \frac{p_0}{\rho_0(\gamma-1)} = c_{s0}^2 / \gamma(\gamma-1) = \text{thermal energy/unit mass}$

total thermal energy inside R_{max}

R_{max} is point at which energy of explosion \approx total thermal energy inside R_{max}

Dimensional arguments revisited

If medium has finite temperature problem also defined by $c_{s0} (LT^{-1})$

so can define a characteristic lengthscale $\left(\frac{E}{\rho_0 c_{s0}^2}\right)^{1/3}$ that is of order R_{max} and timescale $\frac{1}{c_{s0}} \left(\frac{E}{\rho_0 c_{s0}^2}\right)^{1/3}$

for $T_0 \approx 10^4 K$, $R_{max} \approx 100 pc$, $t_{max} \approx 10 Myr$

As SN rate/vol is $\approx 10^{-7} Myr^{-1} pc^{-3}$ we'd expect a fraction $\frac{4\pi}{3} R_{max}^3 t_{max} SN_{rate} \approx 10$ (ie all) of ISM to be in SN driven bubbles and heated to high temp ($\approx 10^6 K$)

This isn't the case as gas behind shock cools after 0.1 Myr (20 pc) and bubble grows more slowly

Finite scale height of galactic disk \rightarrow bubble blows out of plane and cavity is depressurised

\rightarrow NB this means that only 1% of SN energy is deposited in ISM, rest is radiated away, put into galactic halo etc.

Bernoulli's equation

Applies to steady barotropic (inviscid) flows

Derivation:

Remember inviscid mom. eq. $-\nabla S \frac{dp}{\rho}$ for a barotropic e.o.s.

$$\frac{d\mathbf{u}}{dt} + \mathbf{u} \cdot \nabla \mathbf{u} = -\frac{1}{\rho} \nabla p - \nabla \Phi$$

or if steady $\nabla \left(\frac{1}{2} u^2 \right) = \mathbf{u} \cdot (\nabla \mathbf{u})$ by identity (see Eq. 1.5)

Define vorticity $\boldsymbol{\omega} = \nabla \wedge \mathbf{u}$ (which is a measure of angular velocity of fluid)

$$\mathbf{u} \wedge \boldsymbol{\omega} = \nabla \left[\frac{1}{2} u^2 + S \frac{dp}{\rho} + \Phi \right]$$

Take $\mathbf{u} \cdot$ this eqn gives

$$\mathbf{u} \cdot \nabla H = 0 \quad \text{ie convective derivative of H is zero!}$$

where $H = \frac{1}{2} u^2 + S \frac{dp}{\rho} + \Phi = \text{Bernoulli const}$
 = constant along streamlines

$$DH/dt = \frac{\partial H}{\partial t} + \mathbf{u} \cdot \nabla H = 0$$

Meaning

If $p=0$, just says that KE + PE = const.

If $\Phi = \text{const}$, kinetically it describes how KE is converted betw. random molecular motion (p) and bulk flow (u)

hydrodynamically it reflects fact that pressure differences are req. to accelerate the flow.

EA Aeroplane wing: curved $\rightarrow u$ higher on top $\rightarrow p$ lower \rightarrow lift
 Shower curtain: downward vel of air when turned on $\rightarrow p$ lower
 Water draining from tank etc.

Irrotational flows $\rightarrow \boldsymbol{\omega} = 0$ (also called "curl-free")

In this case $\nabla H = 0$, so H is constant everywhere (not just on streamlines)

Also means it's possible to write $\mathbf{u} = -\nabla \Phi_u$ (as $\nabla \wedge \nabla \Phi_u = 0$ for all Φ)

eg Stokes theorem $\int_S \nabla \wedge \mathbf{u} \cdot d\mathbf{S} = \oint_C \mathbf{u} \cdot d\mathbf{l}$

So $\oint_C \mathbf{u} \cdot d\mathbf{l} = 0$ for irrotational flow, and a uniform flow is, but a rotating flow is not.

Incompressible flows $\rightarrow \nabla \cdot \mathbf{u} = 0$ (from 2nd lecture)

This means it's possible to write $\mathbf{u} = \nabla \wedge \phi$ (as $\nabla \cdot \nabla \wedge \mathbf{u} = 0$ for all \mathbf{u})

eg for 2D flow $\mathbf{u} = -\nabla \wedge [\phi(x, y) \hat{\mathbf{e}}_z]$

$$\therefore u_x = -\partial \phi / \partial y \quad \text{and} \quad u_y = \partial \phi / \partial x$$

and for an irrotational incompressible flow: $\nabla^2 \Phi_u = 0$

De Laval Nozzle (linear application of Bernoulli's eqn);

Consider steady barotropic (inviscid) flow in z dir= down tube of variable cross-section A(z), and ignore gravity.

Continuity $\rightarrow \rho u A = \text{const} = \dot{M}$ (mass flow rate)
 Momentum $\rightarrow \frac{d}{dt} \int_V \rho \mathbf{u} dV + \mathbf{u} \cdot \nabla \mathbf{u} = -\frac{1}{\rho} \nabla p - \nabla \Phi$ (as steady, ignore gravity)
 $= -c_s^2 \frac{1}{\rho} \nabla \rho$ (as barotropic)
 $= -c_s^2 \nabla \ln \rho$

Flow is uniform so irrotational $\therefore \mathbf{u} \cdot \nabla \mathbf{u} = \nabla(\frac{1}{2} u^2) - \mathbf{u} \wedge (\nabla \wedge \mathbf{u})$
 $= u \nabla u = u^2 \nabla \ln u$

But continuity $\rightarrow \ln \rho + \ln u + \ln A = \ln \dot{M}$
 $\therefore \nabla \ln \rho = -\nabla \ln u - \nabla \ln A$

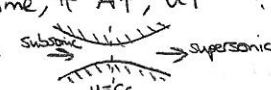
Combining $\rightarrow (u^2 - c_s^2) \nabla \ln u = c_s^2 \nabla \ln A$

gas can only make a sonic transition (eg sub- to super-sonic) at a max or min in A but a max or min in A could also correspond to a max or min in u *

Bernoulli's eq w/o gravity: $\frac{1}{2} u^2 + \int \frac{dp}{\rho} = \frac{1}{2} u^2 + \int \frac{c_s^2}{\rho} dp = \text{const.}$

If isothermal $c_s^2 = \frac{R}{M} T = \text{const}$
 $\therefore \frac{1}{2} u^2 + c_s^2 \ln \rho = \text{const.}$
 If there's a sonic trans at A_m , that const is $c_s^2 (\frac{1}{2} + \ln \rho_m)$
 $\therefore u^2 = c_s^2 [1 + 2 \ln(\frac{\rho}{\rho_m})]$
 $= c_s^2 [1 + 2 \ln(\frac{u A}{c_s A_m})]$ thus giving $u(z)$ and $\rho(z)$ for given $A(z), c_s, \dot{M}$

If polytropic $p = K \rho^{1+\frac{1}{n}}$
 $c_s^2 = (\frac{n+1}{n}) K \rho^{\frac{1}{n}}$
 $\int \frac{c_s^2}{\rho} dp = n c_s^2$
 If there's a sonic trans at $A_m, \dot{M} = \rho_m c_{sm} A_m = K^{1/2} (\frac{n+1}{n})^{1/2} A_m \rho_m^{1+\frac{1}{2n}}$
 $\therefore \rho_m = \left[\left(\frac{\dot{M}}{A_m} \right)^2 \frac{n}{K(n+1)} \right]^{\frac{n}{2n+1}}$
 Put into Bernoulli's eq $\rightarrow \frac{1}{2} \left(\frac{\dot{M}}{\rho A} \right)^2 + (n+1) K \rho^{\frac{1}{n}} = c_{sm}^2 (\frac{1}{2} + n)$
 $= (\frac{1}{2} + n) (\frac{n+1}{n}) K \rho_m^{1/n}$
 giving $\rho(z)$ and $u(z)$ for given $\dot{M}, A(z), K, n$

* In subsonic regime, if $A \downarrow, u \uparrow$ (eg rivers flowing through narrows);
 supersonic regime, if $A \uparrow, u \uparrow$ (as supersonic flows are compressible $\rightarrow \rho \downarrow$ and $u \uparrow$ to keep \dot{M} const);
 eg: Jet Engine  \rightarrow supersonic
 \hookrightarrow NB from above $\nabla \ln \rho / \nabla \ln u = -u^2 / c_s^2$

EC Jets from AGN or PPDs.
 Disk gas \rightarrow easiest to escape \perp to disk \rightarrow acts like pipe, but confinement is through pressure balance
 (These real jets are time variable and include shocks, and probably involve mag. fields for their acceleration and confinement);

Spherical Accretion (3D application analogous to DL nozzle);

Consider point mass star, mass M, accreting gas spherically symmetrically in a steady barotropic inviscid flow.

Gas is accelerated from rest (subsonic) to freefall (supersonic) → must be sonic transition

Continuity → $\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0$
 $\frac{1}{r^2} \frac{\partial}{\partial r} (r^2 \rho u) = 0$ (as spherically symmetric, NB $\mathbf{u} = +u \hat{r}$);
 $4\pi r^2 \rho u = -\dot{M}$

Momentum → $\frac{\partial \rho \mathbf{u}}{\partial t} + \mathbf{u} \cdot \nabla \mathbf{u} = -\frac{1}{\rho} \nabla p - \nabla \Phi$ where $\Phi = -GM/r$
 $u^2 \frac{d \ln u}{dr} = -c_s^2 \frac{d \ln \rho}{dr} - GM/r^2$ (for same reasons as DL nozzle, noting that $\nabla \rightarrow \frac{d}{dr}$ and $\nabla \cdot \mathbf{u} = 0$ for spherically symmetry)

From continuity $\ln \rho + \ln u + \ln r^2 = -\ln \dot{M} / 4\pi$
 $d \ln \rho / dr = -d \ln u / dr - 2/r$
 $\therefore (u^2 - c_s^2) \frac{d \ln u}{dr} = \frac{2c_s^2}{r} \left[1 - \frac{GM}{2c_s^2 r} \right]$

∴ gas can only make a sonic transⁿ at $r_s = \frac{GM}{2c_s^2}$
 but this radius could also correspond to a min or max in u

Bernoulli's eq: $\frac{1}{2} u^2 + \int \frac{c_s^2}{\rho} d\rho - \frac{GM}{r} = \text{const}$

If isothermal (Bondi accretion)

Given T, know c_s (which is const) and so given M, r_s
 But not ρ_s or \dot{M} that are related through $\dot{M} = 4\pi r_s^2 \rho_s c_s$ (assuming u is -ve)
 Bernoulli → $\frac{1}{2} u^2 + c_s^2 \ln \rho - GM/r = \frac{1}{2} c_s^2 + c_s^2 \ln \rho_s - GM/r_s$
 $\therefore u^2 = 2c_s^2 \left[\ln \rho / \rho - 3/2 \right] + 2GM/r$

As $r \rightarrow 0$, $u \rightarrow \sqrt{2GM/r}$ (free-fall)
 $r \rightarrow \infty$, we know $u \rightarrow 0$ so $\rho_s = \rho_{\infty} e^{3/2}$ and $\dot{M} = \pi (GM)^2 \rho_{\infty} e^{3/2} / c_s^3$
 So problem well defined given T, M, ρ_{∞}

EC: $1 M_{\odot}$ in $200K$, 10^6 H atoms/ m^3 ISM → $r_s = 4 \times 10^{13} m$, $\dot{M} = 3 \times 10^{18} \text{ kg/yr}$ ($0.01 M_{\odot} / 10 \text{ Gyr}$)
 → Stars accrete from ISM that is $10^6 \times$ denser, where self-gravity important

If polytropic

Bernoulli → $\frac{1}{2} u^2 + n c_s^2 - \frac{GM}{r} = c_s^2 \left[\frac{1}{2} + n - 2 \right]$ (evaluating at sonic transⁿ)
 $\frac{1}{2} u^2 + (n+1) k \rho^{-1/n} - \frac{GM}{r} = (n - \frac{3}{2}) c_s^2$ (substituting $c_s^2 = \left(\frac{n+1}{n}\right) k \rho^{1/n}$)

As $r \rightarrow \infty$, $u \rightarrow 0$, so $(n - \frac{3}{2}) c_s^2 = (n+1) k \rho_{\infty}^{1/n}$
 ∴ if $n < 3/2$ the sonic point is never reached (because gas too incompressible and pressure gradient prevents flow being accelerated)

Now $\dot{M} = 4\pi r_s^2 \rho_s c_s = \pi (GM)^2 \rho_s c_s^{-3}$ (just substituting in for r_s)
 but $\rho_s^{1/n} = \left(\frac{n}{n+1}\right) k^{-1} c_s^2$ (from same $c_s^2 = \dots$ eqn^s above)
 $\therefore \dot{M} = \pi (GM)^2 \left(\frac{n}{n+1}\right)^n k^{-n} c_s^{2n-3}$ (and so \dot{M} is given in terms of M, k, n, ρ_{∞})

NB, if $n = 3/2$, which is adiabatic flow of monatomic gas, $c_s \rightarrow \infty$ and so $r_s \rightarrow 0$
 but $\dot{M} \rightarrow \pi (GM)^2 \left(\frac{2}{3}\right)^{3/2} k^{-3/2}$

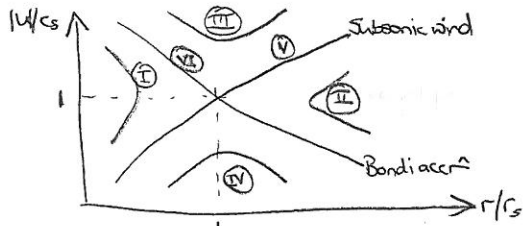
Finally $\frac{1}{2} \left[\frac{\dot{M}}{4\pi r^2 \rho} \right]^2 + (n+1) k \rho^{1/n} = (n+1) k \rho_{\infty}^{1/n} + GM/r$ gives ρ vs r

Stellar winds

Equations are exactly same as for accretion, except u is +ve and \dot{M} is -ve
 But here b.c. are set at inner boundary

If isothermal $\Rightarrow (u^2 - c_s^2) d \ln u / dr = \frac{2c_s^2}{r} [1 - \frac{GM}{2c_s^2 r}]$

$\therefore (\frac{u}{c_s})^2 - \ln(\frac{u}{c_s})^2 = 4 \ln(\frac{r}{r_s}) + 4(\frac{r_s}{r}) + C$



Nature of sfs dep on C:

I and II are unphysical (two values of u at each r)
 (flow w such "C" would readjust to new st. st.)

III and IV are always sub or supersonic

V and VI have $C = -3$

NB st^c is steady, but winds can be variable (eg CME), and note that interacts w ISM in shock

Astrophysical importance:

- source of mechanical energy to ISM $\sim \frac{1}{2} \dot{M}_{wind} V_{\infty}^2$ (terminal velocity)
- (typically comparable to SN input, but dominates in < 2 Myr clusters)
- returns products of nucleosynthesis to ISM

Drivers

OB^{*}s: rad= absorbed by metallic ions
 10^{-6} to 10^{-4} M_⊙/yr, $V_{\infty} \approx 1000$ km/s

AGB: rad= absorbed by dust that forms in wind flow
 10^{-8} to 10^{-4} M_⊙/yr, $V_{\infty} \approx 20$ km/s

Vorticity evolution

Helmholtz's equation

Rewrite barotropic inviscid mom. eq:

$$\partial \underline{u} / \partial t = \underline{u} \wedge \underline{\omega} - \nabla H$$

Take curl of this eq.

$$\partial \underline{\omega} / \partial t = \nabla \wedge (\underline{u} \wedge \underline{\omega}) \quad \text{as } \nabla \wedge \nabla H = 0!$$

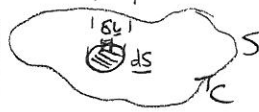
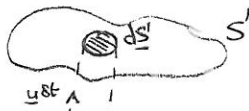
So an irrotational flow ($\underline{\omega} = 0$) remains as such ... (NB $\partial \underline{\omega} / \partial t = \nabla \wedge (\underline{u} \wedge \underline{\omega}) + \underline{u} \wedge \partial \underline{\omega} / \partial t$)
 ... unless flow is viscous in which case vorticity can be created

Kelvin's vorticity theorem

Consider vorticity linked to surface S inside a fluid $\int_S \underline{\omega} \cdot d\underline{S}$

This can change due to intrinsic change in $\underline{\omega}$ or change in surface S caused by flow

$$D \int_S \underline{\omega} \cdot d\underline{S} / Dt = \int_S \partial \underline{\omega} / \partial t \cdot d\underline{S} + \int_S \underline{\omega} \cdot D d\underline{S} / Dt$$



Consider area element that changed from $d\underline{S}$ to $d\underline{S}'$ in time δt , and the volume V to $d\underline{S}$ and $d\underline{S}'$ at ends.

If $d\underline{l}$ is length element on curve bounding $d\underline{S}$, the vector area of side of this volume is $\oint \delta t \underline{u} \wedge d\underline{l}$

But $d\underline{S}' - d\underline{S} - \oint \delta t \underline{u} \wedge d\underline{l} = 0$ (sum of outwardly pointing vector areas in V)

$$\therefore D d\underline{S} / Dt = \oint \underline{u} \wedge d\underline{l}$$

$$\text{Thus } \int_S \underline{\omega} \cdot D d\underline{S} / Dt = \int_S \oint \underline{\omega} \cdot (\underline{u} \wedge d\underline{l})$$

$$= \int_S \oint (\underline{\omega} \wedge \underline{u}) \cdot d\underline{l} \quad (\text{as } \underline{a} \cdot (\underline{b} \wedge \underline{c}) = \underline{c} \cdot (\underline{a} \wedge \underline{b}))$$

$$= \oint_C (\underline{\omega} \wedge \underline{u}) \cdot d\underline{l} \quad (\text{as all inner components cancel})$$

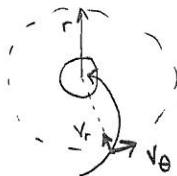
$$= \int_S \nabla \wedge (\underline{\omega} \wedge \underline{u}) \cdot d\underline{S} \quad (\text{by Stokes' theorem})$$

$$\therefore D \int_S \underline{\omega} \cdot d\underline{S} / Dt = \int_S d\underline{S} \cdot [\partial \underline{\omega} / \partial t - \nabla \wedge (\underline{u} \wedge \underline{\omega})] = 0 \quad (\text{by Helmholtz' eq.})$$

\therefore flux of vorticity is conserved and moves w fluid (that is barotropic and inviscid)

$$\therefore D \oint_C \underline{u} \cdot d\underline{l} / Dt = 0$$

eg. bath tub vortex



$$\text{circulation } \Gamma = 2\pi r v_\theta = \text{const}$$

$$v_\theta \propto 1/r$$

• tornado has same effect, noting also that as $v_\theta \uparrow$ $p \downarrow$

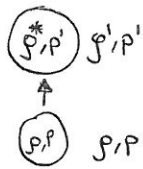
Fluid instabilities

Consider a steady fluid ($\partial/\partial t = 0$)

If small perturb grow w time, that configuration is said to be unstable w.r.t. pert \Rightarrow (unlikely found in nature)
 A stable config. \rightarrow pert-ns diminish, or oscillate (eg. sound waves)

Convective instability

Consider perfect gas in hydrostatic \Rightarrow in uniform grav. field in $-\hat{z}$ dir



Take fluid element at same p, ρ as surroundings
 Displace by δz where local cond's are $p' = p + \frac{dp}{dz} \delta z$, $\rho' = \rho + \frac{d\rho}{dz} \delta z$
 Pressure imbalances quickly removed by acoustic waves, but heat exchange takes longer \rightarrow gas changes adiabatically to new p', ρ^*

As $p/\rho^\gamma = p'/\rho'^\gamma \rightarrow \rho^* = \rho (p'/p)^{1/\gamma} = \rho + \left(\frac{\rho}{\gamma p}\right) \frac{dp}{dz} \delta z$ (to 1st order)

Archimedes' principle \rightarrow if $\rho^* < \rho'$ gas continues to rise (but sinks if $\rho^* > \rho'$)

Schwarzschild criterion \rightarrow instability if $\left(\frac{\rho}{\gamma p}\right) \frac{dp}{dz} < \frac{d\rho}{dz}$

eg. if surrounding were adiabatic (p/ρ^γ const) \rightarrow neutrally stable
 but if p/ρ^γ decreases w $z \rightarrow$ unstable

Alternatively, as $p = \frac{\rho R}{M} T \rightarrow dp/dz = \frac{\rho}{T} \frac{dT}{dz} - \frac{\rho}{T^2} \frac{dT}{dz}$

$\therefore \rho^* - \rho' = \left[-\left(1 - \frac{1}{\gamma}\right) \frac{\rho}{p} \frac{dp}{dz} + \frac{\rho}{T} \frac{dT}{dz} \right] \delta z$

Then as dp/dz and dT/dz have same sign (from ideal gas);
 and $dp/dz < 0$ (from momentum eq. - pressure supports gravity);

\rightarrow system stable if $\left| \frac{dT/dz}{T} \right| < \left(1 - \frac{1}{\gamma}\right) \frac{1}{p} \left| dp/dz \right|$

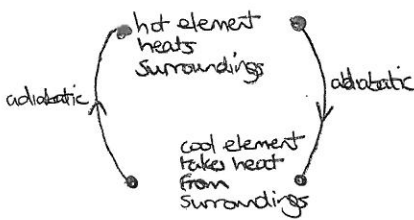
If stable \rightarrow internal gravity waves

e.o.m. of fluid element: $\rho^* \frac{d^2 \delta z}{dt^2} = -(\rho^* - \rho') g$

$\therefore \frac{d^2 \delta z}{dt^2} + \frac{g}{T} \left[\frac{dT}{dz} - \left(1 - \frac{1}{\gamma}\right) \frac{1}{p} dp/dz \right] \delta z = 0$

\rightarrow SHM w angular freq. $\sqrt{\frac{g}{T} \left[\frac{dT}{dz} - \left(1 - \frac{1}{\gamma}\right) \frac{1}{p} dp/dz \right]}$

If unstable \rightarrow convection cells



size of convective cell set by scale elements cease to be adiabatic (ie exchange heat w surroundings)
 \rightarrow convection transports heat upwards in displaced elements.

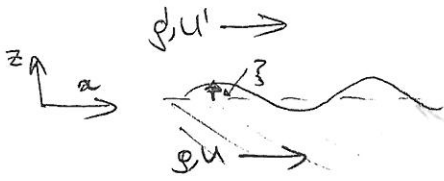
eg: stars in hydrostatic \Rightarrow transport energy by rad unless convectively unstable which occurs in regions of large temperature gradients such as envelopes of low mass stars (caused by high opacity of partially ionised composition due to low T)

Rayleigh-Taylor Instability

(38)

An unstably stratified config. of fluid layers turns over due to gravity

- ⊙ Consider 2 barotropic, ^{inviscid} fluids in uniform ^(vertically down) grav. field w interface at $z=0$, both w uniform flow in x dir:



Assume flow is irrotational \rightarrow can write $\underline{u} = -\nabla\Phi$ (ie. $\Phi = -Ux$);

\therefore Eulerian mom. eq. $\partial(-\nabla\Phi)/\partial t + \nabla(\frac{1}{2}u^2) = -\nabla(\frac{p}{\rho}) - \nabla\Phi$

Also assume incompressible, and note $\Phi = gz$

$$\therefore -\nabla\partial\Phi/\partial t + \nabla\frac{1}{2}u^2 + \nabla\frac{p}{\rho} + \nabla gz = 0$$

Integrate \rightarrow

$$-\partial\Phi/\partial t + \frac{1}{2}u^2 + p/\rho + gz = F(t) \quad (\text{const in space, but not nec. int.})$$

Note that uniform flow is steady solⁿ w F as Bernoulli's const., and that there's a similar eqⁿ w dashes for top fluid

- ⊙ Consider a pertⁿ to interface $\zeta(x,t)$

w corresponding velocity pertⁿ $\Phi = -Ux + \phi$ and $\Phi' = -U'x + \phi'$

s.t. $u^2 = |\nabla\Phi|^2 = U^2 - 2U\partial\phi/\partial x$ (to first order)

Use mom. eq. to determine pressure at interface ($z=\zeta$) which is same top and bottom

$$\rightarrow p|_{z=\zeta} = \rho F(t) - \rho \left[-\partial\phi/\partial t|_{z=\zeta} + \frac{1}{2}u^2 - U\partial\phi/\partial x|_{z=\zeta} + g\zeta \right] = p'|_{z=\zeta}$$

$$\therefore \rho \left[-\partial\phi/\partial t|_{z=\zeta} + \frac{1}{2}u^2 - U\partial\phi/\partial x|_{z=\zeta} + g\zeta \right] - \rho' \left[-\partial\phi'/\partial t|_{z=\zeta} + \frac{1}{2}u'^2 - U'\partial\phi'/\partial x|_{z=\zeta} + g\zeta \right] = K(t) \quad \text{Ⓐ}$$

Cancellations arise because the constants are same w/o pertⁿ (eg. pertⁿs vanish at $z=\pm\infty$)

Also remember $\nabla^2\Phi = 0$ for irrotational incompressible fluid

$$\therefore \nabla^2\phi = \nabla^2\phi' = 0 \quad \text{Ⓑ}$$

And consider vertical velocity of fluid element just below interface (to 1st order)

$$-\partial\phi/\partial z|_{z=0} = D\zeta/Dt = \partial\zeta/\partial t + U\partial\zeta/\partial x \quad \text{Ⓒ}$$

$$\text{and } -\partial\phi'/\partial z|_{z=0} = \partial\zeta/\partial t + U'\partial\zeta/\partial x$$

- ⊙ Seek solutions of form $\zeta = Ae^{i(kx-wt)}$

$$\phi = Ce^{i(kx-wt)+jz} \quad \text{and } \phi' = C'e^{i(kx-wt)+j'z}$$

Sub into

$$\text{Ⓑ } -k^2 + j^2 = -k^2 + j'^2 = 0$$

For $\phi \rightarrow 0$ at $z \rightarrow -\infty \rightarrow j = +k$

likewise $j' = -k$

$$\text{Ⓒ } -kC = i(kU-w)A$$

$$kC' = i(kU'-w)A$$

$$\text{Ⓐ } \rho(-i(kU-w)C + gA) = \rho'(-i(kU'-w)C' + gA)$$

(iv) 3 eqns, 3 unknowns (A, C, C')

eliminate unknowns to get dispersion rel =

$$\rho(ku - \omega)^2 + \rho'(ku' - \omega)^2 = kg(\rho - \rho')$$

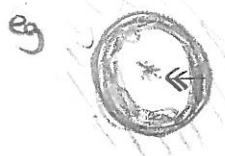
$$-(\omega/k)^2(\rho + \rho') - 2(\frac{\omega}{k})(\rho u + \rho' u') + [\rho u^2 + \rho' u'^2 + \frac{g}{k}(\rho - \rho')] = 0$$

$$\therefore \text{phase velocity of wave } \omega/k = \left[\frac{\rho u + \rho' u'}{\rho + \rho'} \right] \pm \sqrt{\frac{g}{k} \frac{\rho - \rho'}{\rho + \rho'} - \frac{\rho \rho' (u - u')^2}{(\rho + \rho')^2}}$$

If $u = u' = 0$

and $\rho' < \rho \rightarrow \omega/k = \pm \sqrt{\frac{g}{k} \frac{\rho - \rho'}{\rho + \rho'}}$
 Surface gravity waves w/ vel. that dep on k (ie. dispersive)

$\rho' > \rho \rightarrow \omega/k = \pm i \sqrt{\frac{g}{k} \frac{\rho - \rho'}{\rho + \rho'}}$
 $\therefore \xi = A e^{ikx} e^{\sqrt{\frac{g}{k}(\rho - \rho')/(\rho + \rho')} t}$ is exponentially growing mode (RT instab.)
 NB a light fluid accelerating into a heavy fluid (w/o gravity) gives same instab.



eg Blast wave \rightarrow thin shell that is decelerating
 \rightarrow in the frame of interface the fluids experience outward acceler \rightarrow dense (shell) on top of tenuos (ISM)
 \rightarrow RT instab \rightarrow filaments in shell.

If $u \neq 0, u' \neq 0$ and $\rho' > \rho$ \rightarrow Kelvin Helmholtz instability \rightarrow RT stable

Unstable if term in square root is -ve (as \rightarrow imaginary part of $\omega \rightarrow$ exponential growth)

$$\rightarrow \frac{\rho \rho' (u - u')^2}{k} > \left(\frac{g}{k}\right) (\rho^2 - \rho'^2)$$

$$\text{or } k > (\rho^2 - \rho'^2)g / \rho \rho' (u - u')^2$$

\rightarrow Interface betw two fluids w/vels if they move at diff speeds and pert \rightarrow large enough k (small enough wavelength) are always unstable (unless damped by surface tension)

eg. jet moving w.r.t. ISM or ICM (esp as $g \neq 0$)

Gravitational (Jeans) Instability

Same analysis as for sound waves ^{in uniform medium}, but including self gravity (→ extra term in mom eq, add Poisson's eq, and include $\Delta\Phi$)

(c) $\partial\rho/\partial t + \nabla \cdot (\rho\mathbf{u}) = 0$
 $\partial\mathbf{u}/\partial t + \mathbf{u} \cdot \nabla\mathbf{u} = -\frac{1}{\rho}\nabla p - \nabla\Phi$
 $\nabla^2\Phi = 4\pi G\rho$

(b) Equilibrium $\mathbf{u} = 0, p = p_0, \rho = \rho_0, \Phi = \Phi_0$?
 Not quite: hydrostatic and uniform → $\frac{1}{\rho_0}\nabla p_0 = -\nabla\Phi_0$ and $\nabla^2\Phi_0 = 4\pi G\rho_0$
 this is zero ... so, so is this ... and this ... so a static Universe is empty!
 But ~~estimate~~ → Jeans's swindle (NB, should perturb hydrostatic isothermal slab solⁿ)

(c) Perturb: $\mathbf{u} = \Delta\mathbf{u}, p = p_0 + \Delta p, \rho = \rho_0 + \Delta\rho, \Phi = \Phi_0 + \Delta\Phi$
 Uniform, so $\delta X = \Delta X$

(d) $\partial\Delta\rho/\partial t + \rho_0 \nabla \cdot \Delta\mathbf{u} = 0$
 $\partial\Delta\mathbf{u}/\partial t = -c_s^2 \frac{1}{\rho_0} \nabla \Delta\rho - \nabla \Delta\Phi$ ($c_s^2 = dp/d\rho$ i.e. assuming barotropic)
 $\nabla^2 \Delta\Phi = 4\pi G \Delta\rho$

(e) Let $\Delta\rho = \rho_1 e^{i(\mathbf{k}\cdot\mathbf{x} - \omega t)}, \Delta\Phi = \Phi_1 e^{i(\mathbf{k}\cdot\mathbf{x} - \omega t)}, \Delta\mathbf{u} = \mathbf{u}_1 e^{i(\mathbf{k}\cdot\mathbf{x} - \omega t)}$

$-\rho_1 \omega + \rho_0 \mathbf{k} \cdot \mathbf{u}_1 = 0$
 $-\omega \mathbf{u}_1 = -c_s^2 \frac{\rho_1}{\rho_0} \mathbf{k} - \Phi_1 \mathbf{k}$
 $-\mathbf{k}^2 \Phi_1 = 4\pi G \rho_1$

3 eqs, 3 unknowns ($\mathbf{u}_1, \Phi_1, \rho_1$) to be eliminated to get dispersion relⁿ:

$\omega^2 = c_s^2 (\mathbf{k}^2 - \mathbf{k}_J^2)$
 where $\mathbf{k}_J^2 = 4\pi G \rho_0 / c_s^2$

∴ unstable if $\mathbf{k}_J^2 > \mathbf{k}^2$ (as then ω is imaginary → exponentially growing mode)

[In normal sound waves a compression is counteracted by pressure; here the self gravity of the compression makes density of compression increase]

∴ Unstable if $\lambda = \frac{2\pi}{k} > 2\pi/k_J = \sqrt{\frac{\pi c_s^2}{G \rho_0}} = \text{Jeans's length}$

* The mass within this wavelength $\sim \rho_0 (2\pi/k_J)^3 \sim \frac{\pi^{3/2} c_s^3}{G^{3/2} \rho_0^{1/2}} = \text{Jeans's mass}$

As long wavelength that evolves slowly → wave isothermal: $c_s^2 \sim \frac{P^*}{\rho} T$

→ Jeans's mass is $\sim \left(\frac{\pi P^* T}{G} \right)^{3/2} \rho_0^{-1/2}$

* NB This lengthscale is that over which sound crossing time L/c_s is equal to free-fall time due to self gravity $L/\sqrt{G\rho_0} \sim L^2/L$

[→ sound waves are slower than collapse and so can't set up pressure gradients to counteract]

Astrophysically explains why Universe not smooth

→ star form = (in GMCs $T \sim 10\text{K}, n = 10^{21}\text{m}^{-3} \rightarrow M_{\odot} \sim 1 M_{\odot}$)

galaxy form = (densities too low in current LCM → galaxies formed when Universe younger and denser)

planet form = (maybe)

Thermal Instability

Runaway heating (or cooling) following pertⁿ ΔT from initial state of thermal = \dots

Simple analysis: If gas heats up at const p , then $\dot{Q} \Rightarrow \dot{Q} + \left(\frac{\partial \dot{Q}}{\partial T}\right)_p \Delta T$
So unstable if $(\partial \dot{Q} / \partial T)_p < 0$ (NB \dot{Q} is rate of cooling);

Full pertⁿ analysis

Assume uniform medium, ignore gravity, so perturbed mass and momentum eqs are:

$$\frac{\partial \Delta \rho / \partial t + \rho_0 \nabla \cdot \Delta u}{\partial \Delta u / \partial t} = -\frac{1}{\rho_0} \nabla \Delta p \quad (\text{NB have kept } \Delta p \text{ here});$$

To get energy eqn here, first write $p = K \rho^\gamma$ where K is a variable (i.e. not necessarily adiabatic)

$$\therefore dp = \rho^\gamma dK + \gamma \frac{p}{\rho} d\rho = \frac{p}{\rho} d\rho + \left(\frac{p}{K}\right) p dT \quad (\text{from ideal gas law})$$

$$\rho^\gamma dK = \frac{p}{\rho} (1-\gamma) d\rho + \left(\frac{p}{K}\right) p dT$$

$$dK = \rho^{1-\gamma} (1-\gamma) \left[\frac{p}{\rho} d\rho - \left(\frac{R_p/M}{\gamma-1}\right) dT \right]$$

From 1st law of thermodynamics, $\text{heat} = -dQ$ ($dQ = \text{heat}$ done)

$$\therefore dK/dt = (1-\gamma) \rho^{-\gamma} \dot{Q} \quad (\text{NB similar eqn derived to consider entropy change across shock})$$

So for perturbations:

$$d \Delta K / dt = -(\gamma-1) \rho_0^{1-\gamma} \left[\left(\frac{\partial \dot{Q}}{\partial p}\right)_p \Delta p + \left(\frac{\partial \dot{Q}}{\partial \rho}\right)_p \Delta \rho \right]$$

$$= -A^* \Delta p - B^* \Delta \rho$$

And $\Delta p = \rho_0^\gamma \Delta K + (\gamma p_0 / \rho_0) \Delta \rho$

(Note if we assume adiabatic s.t. $\Delta K = 0$, or isothermal $dT=0$, we get normal sound wave solⁿ)

• Now let $\Delta \rho = \rho_1 e^{i(k \cdot x) + qt}$, $\Delta u = u_1 e^{i(k \cdot x) + qt}$, $\Delta K = K_1 e^{i(k \cdot x) + qt}$, $\Delta p = p_1 e^{i(k \cdot x) + qt}$
 \rightarrow unstable if q is real and +ve (if imaginary get oscillatory solⁿ)

$$\therefore q \rho_1 + \rho_0 i k \cdot u_1 = 0$$

$$q u_1 = -\frac{1}{\rho_0} i k p_1$$

$$q K_1 = -A^* p_1 - B^* \rho_1$$

$$p_1 = \rho_0^\gamma K_1 + (\gamma p_0 / \rho_0) \rho_1$$

4 eqns, 4 unknowns (ρ_1, p_1, K_1, u_1) to be eliminated to get dispersion relⁿ:

$$\rightarrow E(q) \equiv q^3 + (A^* \rho_0^\gamma) q^2 + (k^2 \gamma p_0 / \rho_0) q - B^* k^2 \rho_0^\gamma = 0 \quad (\text{is a polynomial in } q)$$

As $E(\infty) = \infty$ and $E(0) = -B^* k^2 \rho_0^\gamma$, there is a real +ve root (\rightarrow unstable) if $B^* > 0$
i.e. if $(\partial \dot{Q} / \partial p)_p = -\left(\frac{p}{\rho}\right) (\partial \dot{Q} / \partial T)_p > 0$

$$\rightarrow (\partial \dot{Q} / \partial T)_p < 0 \Rightarrow \text{Field criterion (for instability)}$$

Same as simple analysis! But, could also be unstable if $B^* < 0$ depending on A^* (must be large and -ve), though field criterion is good determinant of thermal stability

* Eg. If $\dot{Q} = A_p T^\alpha - H$ (cooling processes, eg const heating by cosmic rays)
 $= (A_p M / R \epsilon) T^{\alpha-1} - H$

then field criterion says unstable if $\alpha < 1$ (such as optically thin thermal Bremsstrahlung, $\alpha = 1/2$)

• Explains temp structure of ISM, as warm 10^4 K and cold 10^2 K neutral atomic phases coexist (along w hotter phase ionised by SN), which is only poss. if gas is thermally unstable

* NB growth timescale just given by $1/q$ and depends on k , and in various limits is just balance betw two of five terms. Eg for short wavelength (large k) it's last 2 terms so $q \sim B^* \rho_0^\gamma / (\gamma p_0 / \rho_0)$.
If cooling mostly at const p , $B^* \rho_0^\gamma \propto \Delta \dot{Q}$ and $\gamma p_0 / \rho_0 \propto E_0 \rightarrow$ this is the thermal timescale

Viscous flows

Remember! Eulerian eqn for rate of change of momentum density in component form;

$$\partial \rho u_i / \partial t = -\partial_j \sigma_{ij} + \rho g_i$$

where σ_{ij} = stress tensor

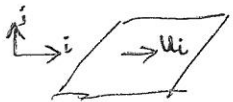
= force per unit area in i dir^s acting on surface \bar{n} normal in j dir^s

$$= \underbrace{\rho u_i u_j}_{\text{ram}} + \underbrace{p \delta_{ij}}_{\text{thermal}} - \underbrace{\sigma'_{ij}}_{\text{viscous stress tensor}}$$

(momentum advected w/ fluid); force due to pressure differentials;

Here, add term due to differential motion of neighbouring fluid elements

Linear shear flow



Consider flow w/ \parallel streamlines in i dir^s

but velocity gradient \perp to streamlines in j dir^s

- appropriate scaling factor
- rms ptcl velocity
- boltzmann's const
- temperature
- ptcl mass

Thermal motion \rightarrow particles have non-zero velocities in j dir^s: $v_j \sim \alpha \sqrt{\frac{RT}{m}}$

momentum flux across surface \bar{n} normal in j dir^s: $\rho u_i v_j$ (both in $+j$ and $-j$ directions)

That momentum is redistributed in particle interactions over scale of mean free path

$$\delta L \sim \frac{1}{\sigma n} = \frac{m}{\pi a^2 \rho} \quad (\text{for hard spheres radius } a)$$

there is also momentum flux across surface from fluid δL below the surface of $\rho(u_i - \delta L \partial_j u_i) v_j$ and similarly \uparrow in the $-j$ direction from fluid δL above the surface;

and net mom flux is: $-\rho \partial_j u_i \alpha \sqrt{\frac{RT}{m}} \frac{m}{\pi a^2 \rho} = -\eta \partial_j u_i$

where $\eta = \left(\frac{5\sqrt{\pi}}{64\pi a^2}\right) \sqrt{mKT} = \text{shear viscosity coefficient}$

$\therefore \sigma'_{ij} = \eta \partial_j u_i$ (as $\partial_j (-\eta \partial_j u_i)$ is change in i th component of momentum due to mismatch in i momentum carried through a cell in each of 3 orthogonal dir^s)

Note: η is independent of ρ (as although $\rho \uparrow \rightarrow$ mom flux \uparrow , also $\delta L \downarrow$); but does depend on T (as $T \uparrow \rightarrow v_j \uparrow$ so mom flux \uparrow);

More general form of σ_{ij} (for molecular viscosity ζ):

$$\text{Let } \sigma_{ij} = \eta (\partial_j u_i + \partial_i u_j - \frac{2}{3} \delta_{ij} \partial_k u_k) + \zeta \delta_{ij} \partial_k u_k$$

$$= \begin{pmatrix} \zeta \partial_k u_k & \eta (\partial_1 u_2 + \partial_2 u_1) & \eta (\partial_1 u_3 + \partial_3 u_1) \\ \eta (\partial_2 u_1 + \partial_1 u_2) & \zeta \partial_k u_k & \eta (\partial_2 u_3 + \partial_3 u_2) \\ \eta (\partial_3 u_1 + \partial_1 u_3) & \eta (\partial_3 u_2 + \partial_2 u_3) & \zeta \partial_k u_k \end{pmatrix}$$

- Note • σ_{ij} is symmetric ($\sigma_{ij} = \sigma_{ji}$) s.t. force on j th face of infinitesimally small cube in i dir is balanced by that on i th face in j dir \rightarrow cube not torqued up (though note that a large cube can be torqued up if velocity field varied across cube);
- diagonal terms are equal, to $\zeta \nabla \cdot \underline{u}$ (ie \propto local rate of change of density); \rightarrow if fluid is squashed, normal forces resisting this are same in all dirs \rightarrow fluid is isotropic
 - ζ = coefficient of bulk viscosity, and this term is assoc. w mom. transfer due to bulk compression of flow (cf. η term assoc. w mom transfer in shear flows)



Navier-Stokes eqn

Mom eq: $\partial_t \rho u_i = -\partial_j \rho u_i u_j - \partial_j p \delta_{ij} + \partial_j [\eta (\partial_j u_i + \partial_i u_j - \frac{2}{3} \delta_{ij} \partial_k u_k) + \zeta \delta_{ij} \partial_k u_k] + \rho g_i$

Using cont. $\rho (\partial_t u_i + u_j \partial_j u_i) = -\partial_i p + \rho g_i + \partial_j [\eta (\partial_j u_i + \partial_i u_j)] - \frac{2}{3} \partial_i (\eta \partial_k u_k) + \partial_i \zeta \partial_k u_k$

Bulk viscosity is important in shocks as causes deceler in dir= normal to shock front, but is of limited relevance elsewhere, so assume $\zeta \rightarrow 0$

Also assume $\eta \rightarrow$ const (eg. isothermal)

$$\rightarrow \rho (\partial_t u_i + u_j \partial_j u_i) = -\partial_i p + \rho g_i + \eta \partial_j \partial_j u_i + \frac{1}{3} \eta \partial_i \partial_j u_j$$

(NB terms combine)

$$\therefore \underline{\partial u} / \partial t + \underline{u} \cdot \nabla \underline{u} = -\frac{1}{\rho} \nabla p - \nabla \Phi + \nu [\nabla^2 \underline{u} + \frac{1}{3} \nabla (\nabla \cdot \underline{u})]$$

where $\nu \equiv \eta / \rho =$ kinematic viscosity

*

Evolution of vorticity

Take curl of NS, noting $\underline{\omega} = \nabla \wedge \underline{u}$, assuming fluid barotropic

$$\partial_t \underline{\omega} - \nabla \wedge (\underline{u} \wedge \underline{\omega}) = \nabla \wedge [\nu \nabla^2 \underline{u} + \frac{1}{3} \nu \nabla (\nabla \cdot \underline{u})]$$

$$= \nu \nabla \wedge [\nabla (\nabla \cdot \underline{u}) - \nabla \wedge (\nabla \wedge \underline{u})] + \nabla \nu \wedge [\nabla^2 \underline{u} + \frac{1}{3} \nabla (\nabla \cdot \underline{u})]$$

If $\nu \ll u$

$$\approx -\nu [\nabla (\nabla \cdot \underline{\omega}) - \nabla^2 \underline{\omega}] + o(\nabla \nu)$$

$$= \nu \nabla^2 \underline{\omega} + o(\nabla \nu)$$

\rightarrow by Kelvin's vorticity theorem, the flux of vorticity is not conserved for viscous fluid

* If $\eta \neq$ const, extra terms on RHS: $\underbrace{(\partial_j \eta) (\partial_j u_i + \partial_i u_j)}_{(\nabla \eta \cdot \nabla) \underline{u}} - \frac{2}{3} \underbrace{(\partial_i \eta) (\partial_k u_k)}_{-\frac{2}{3} \nabla \eta \cdot \nabla \cdot \underline{u}}$

As $\partial_i [(\partial_j \eta) u_j] = \partial_j \eta \partial_i u_j + u_j \partial_i \partial_j \eta$

$$(\partial_j \eta) \partial_i u_j = \nabla (\nabla \eta \cdot \underline{u}) - (\underline{u} \cdot \nabla) \nabla \eta$$

Extra terms on RHS are: $\frac{1}{\rho} [\nabla \eta \cdot \nabla] \underline{u} - \frac{2}{3} \frac{1}{\rho} \nabla \eta \cdot \nabla \cdot \underline{u} + \frac{1}{\rho} \nabla (\nabla \eta \cdot \underline{u}) - \frac{1}{\rho} (\underline{u} \cdot \nabla) \nabla \eta$

Energy dissipation

Consider an incompressible fluid (to make our lives easier!)
 The rate of change of KE is:

$$\begin{aligned} \partial \frac{1}{2} \rho u^2 / \partial t &= u_i \partial \rho u_i / \partial t = -u_i \partial_j \sigma_{ij} + u_i \rho g_i \\ &= -u_i \partial_j \rho u_i u_j - u_i \partial_j \delta_{ij} \rho + u_i \partial_j \sigma'_{ij} + \rho u_i \partial_i \Psi \\ &= -u_i \partial_j \rho u_i u_j - u_i \partial_i \rho + \partial_j u_i \sigma'_{ij} - \sigma'_{ij} \partial_j u_i + \rho u_i \partial_i \Psi \end{aligned}$$

Incompressible $\Rightarrow \partial_i u_i = 0$, and swap indices where both are repeated:

$$\begin{aligned} &= -\partial_j \rho u_j \frac{1}{2} u_i \cdot u_i - \partial_i \rho u_i + \partial_i u_j \sigma'_{ij} - \sigma'_{ij} \partial_j u_i + \partial_i \rho u_i \Psi \\ &= -\partial_i \left[\rho u_i \left(\frac{1}{2} u^2 + \rho / \rho \right) - u_j \sigma'_{ij} \right] - \sigma'_{ij} \partial_j u_i \end{aligned}$$

Integrate over a volume inside fluid (applying divergence theorem $\int \nabla \cdot A dV = \int A \cdot dS$)

$$\partial \int \frac{1}{2} \rho u^2 dV / \partial t = \partial E_K / \partial t = - \underbrace{\int \left[\rho u_i \left(\frac{1}{2} u^2 + \rho / \rho \right) - u_j \sigma'_{ij} \right] \cdot dS}_{\text{advection of KE and PE and internal pressure}} - \underbrace{\int \sigma'_{ij} \partial_j u_i dV}_{\text{viscous stresses integrated over surface}} \quad \text{rate of viscous dissipation in vol.}$$

The first moves energy to neighboring volumes, but the latter is dissipated as heat within vol.

eg choose volume encompassing whole fluid

$$\begin{aligned} \partial E_{tot} / \partial t &= - \int \sigma'_{ij} \partial_j u_i dV \\ &= - \frac{1}{2} \int \sigma'_{ij} (\partial_j u_i + \partial_i u_j) dV \quad \left(\text{as } \sigma'_{ij} = \sigma'_{ji} \right) \\ &= - \frac{1}{2} \int \eta (\partial_j u_i + \partial_i u_j)^2 dV \quad \left(\text{as } \sigma'_{ij} = \eta (\partial_j u_i + \partial_i u_j) \text{ for incompressible!} \right) \end{aligned}$$

$\therefore \eta$ is +ve from 2nd law of thermo.

Ex Incompressible steady flow down circular pipe

45



Continuity and symmetry \rightarrow velocity is only a function of R , not z or θ
 N-S $\rightarrow \underbrace{\frac{du}{dt}}_{\text{steady}} + \underbrace{u \cdot \nabla u}_{\text{by symmetry}} = -\frac{1}{\rho} \nabla p - \underbrace{\nabla \Phi}_{\text{ignore gravity}} + \nu \left[\nabla^2 u + \frac{1}{3} \nabla(\nabla \cdot u) \right]_{\text{incompressible}}$

$$\therefore -\frac{1}{\rho} \nabla p = \nu \nabla^2 u$$

$$\hat{r} \text{ components} \rightarrow \frac{dp}{dR} = \frac{dp}{d\theta} = 0$$

$$\hat{z} \text{ component} \rightarrow \frac{1}{\rho} \frac{dp}{dz} = \nu \frac{1}{R} \frac{d}{dR} (R \frac{du}{dR}) = \text{const.}$$

If pressure changes by Δp over length L : $= \Delta p / \rho L$

Integrating: $u = -\frac{\Delta p}{4\rho \nu L} R^2 + a \ln R + b$

b.c of $u \neq \infty$ @ $R=0 \rightarrow a=0$

$$u=0 \text{ @ } R=R_0 \rightarrow u = \frac{\Delta p}{4\rho \nu L} (R_0^2 - R^2)$$

So mass flow rate $Q = \int_0^{R_0} 2\pi \rho u R dR$
 $= \frac{\pi \Delta p}{8 \nu L} R_0^4$

Turbulence

- \rightarrow flow is irregular in space and time
- \rightarrow break variables into a mean and a fluctuating part
- \rightarrow transition from laminar to turbulent flow depends on Reynolds number, Re
- \rightarrow Re is dimensionless combination of ν ($L^2 T^{-1}$), U (LT^{-1}) and length scale associated w flow L (L)

$$\therefore Re = LU/\nu$$

(transition is at $Re \approx 3000$)

Accretion discs = circular shear flows

eg: circumplanetary, circumstellar, circum-super massive black hole

origin: gas is non zero ang. mom. bound to central object settles into plane defined by mean ang. mom. of gas supply (residual motion damped by shocks);

shear: centrifugal forces balance grav. attraction

$$\rightarrow \Omega^2 R \approx GM/R^2$$

$$\therefore \Omega = \sqrt{GM/R^3} \equiv \text{Keplerian motion}$$

As $d\Omega/dR \neq 0$ viscosity \rightarrow ang. mom. is transferred outward from faster inner regions causing inner material to move in

Equations

Use cylindrical coords and assume axisymmetric ($\partial/\partial\phi = 0$) and negligible vertical motion ($u_z = 0$)

ie $\mathbf{u} = (u_R, u_\phi, 0)$

$$\left. \begin{aligned} \text{NB } \nabla \Psi &= (\partial\Psi/\partial R)\hat{e}_R + (\partial\Psi/\partial z)\hat{e}_z \\ \nabla \cdot \mathbf{A} &= \frac{1}{R} \partial/\partial R [R A_R] + \partial A_z/\partial z \\ \nabla^2 \Psi &= \frac{1}{R} \partial/\partial R [R \partial\Psi/\partial R] + \partial^2 \Psi/\partial z^2 \end{aligned} \right\} \text{ and } (\mathbf{A} \cdot \nabla) \mathbf{B} \text{ for } \partial/\partial\phi = \partial/\partial z = 0$$

$$= (A_R \partial B_R/\partial R - A_\phi B_\phi/R)\hat{e}_R + (A_R \partial B_\phi/\partial R + A_\phi B_R/R)\hat{e}_\phi + A_z \frac{\partial B_z}{\partial z} \hat{e}_z$$

$$\text{and } \nabla^2 \mathbf{u} = (\partial^2 u_R/\partial R^2 + \frac{1}{R} \partial u_R/\partial R - u_R/R^2)\hat{e}_R + (\partial^2 u_\phi/\partial R^2 + \frac{1}{R} \partial u_\phi/\partial R - u_\phi/R^2)\hat{e}_\phi$$

Continuity $\partial\rho/\partial t + \nabla \cdot (\rho \mathbf{u}) = 0$

$$\therefore \partial\rho/\partial t + \frac{1}{R} \partial/\partial R [R \rho u_R] = 0$$

Define $\Sigma = \int \rho dz = \text{surface density}$, and integrate over z

$$\therefore \partial\Sigma/\partial t + \frac{1}{R} \partial/\partial R [R \Sigma u_R] = 0 \quad (1)$$

NS $\rho \mathbf{u} \cdot \nabla \mathbf{u} = -\frac{1}{\rho} \nabla p - \nabla \Psi + \nu \nabla^2 \mathbf{u} + \frac{1}{3} \nu \nabla(\nabla \cdot \mathbf{u}) + \frac{1}{\rho} (\nabla \eta \cdot \nabla) \mathbf{u} + \frac{1}{\rho} \nabla(\nabla \eta \cdot \mathbf{u}) - \frac{1}{\rho} (\mathbf{u} \cdot \nabla) \nabla \eta - \frac{2}{3} \frac{1}{\rho} \nabla \eta \nabla \cdot \mathbf{u}$

For $\partial/\partial\phi = u_z = 0$, and neglecting vertical variation, & u_ϕ , fine of components

$$\therefore \rho \partial u_\phi/\partial t + u_R \rho \partial u_\phi/\partial R + u_\phi u_R \rho/R = \nu [\partial^2 u_\phi/\partial R^2 + \frac{1}{R} \partial u_\phi/\partial R - u_\phi/R^2] + \frac{1}{\rho} \left(\frac{\partial \eta}{\partial R} \right) \left[\frac{\partial u_\phi}{\partial R} - \frac{u_\phi}{R} \right]$$

Integrate over z and define $\langle \nu \rangle = \int \rho \nu dz / \int \rho dz = \Sigma \eta dz / \Sigma$

$$\Sigma (\partial u_\phi/\partial t + u_R \partial u_\phi/\partial R + u_\phi u_R/R) = \langle \nu \rangle \Sigma [\partial^2 u_\phi/\partial R^2 + \frac{1}{R} \partial u_\phi/\partial R - u_\phi/R^2] + \partial[\langle \nu \rangle \Sigma] / \partial R [\partial u_\phi/\partial R - u_\phi/R]$$

Tidying up:

$$\Sigma (\partial u_\phi/\partial t + \frac{u_R}{R} \partial R u_\phi / \partial R) = \frac{1}{R^2} \frac{\partial}{\partial R} [\nu \Sigma R^3 d\Omega/dR] \quad (2)$$

where $\Omega = u_\phi/R$ and we've dropped $\langle \nu \rangle$ around ν

Combine $R u_\phi (1) + R (2)$,

$$\frac{\partial [R \Sigma u_\phi]}{\partial t} + \frac{1}{R} \frac{\partial}{\partial R} [\Sigma R^2 u_\phi u_R] = \frac{1}{R} \frac{\partial}{\partial R} [\nu \Sigma R^3 d\Omega/dR] \quad (3)$$

Eulerian rate of change of ang. mom. / unit area of annulus at R

Net rate of ang. mom. loss from any unit area due to advection of ang. mom. w radial flow

Net viscous torque per unit area

NB viscous stress tensor $\rightarrow \sigma_{r\phi} = \eta R d\Omega/dR = \text{force/area on surface w normal in radial dir}^{\rightarrow}$ in tangential dir (ie from annuli rubbing against e.o.)

$$\therefore \text{torque } G = R \times \sigma_{r\phi} \times 2\pi R 2H$$

$$= 2\pi \nu \Sigma R^3 d\Omega/dR \quad (\text{as } \Sigma = 2H\rho)$$

Different combination of ① and ②

Expect $\partial u_R / \partial t = 0$ (as set by balance of centrifugal and gravitational forces); so ② gives:

$$u_R = \frac{\partial}{\partial R} [v \Sigma R^3 d\Omega / dR] / R \Sigma d(R^2 \Omega) / dR \quad (4)$$

For Keplerian disk $\Omega = \sqrt{GM/R^3}$

$$u_R = -3 \frac{\partial [v \Sigma R^{1/2}]}{\partial R} / \Sigma R^{1/2} \quad (4a)$$

Substitute into ① to get

$$\partial \Sigma / \partial t = \frac{3}{R} \frac{\partial}{\partial R} [R^{1/2} \frac{\partial}{\partial R} [v \Sigma R^{1/2}]] \quad (5) \equiv \text{viscous diffusion eqn}$$

Solution for constant viscosity

Rewrite ⑤ with $s = 2R^{1/2}$

$$\partial \Sigma / \partial t = \frac{12}{s^3} \frac{\partial^2}{\partial s^2} [v \Sigma s]$$

If $v = \text{const}$, then $\partial s / \partial t = 0$

$$\partial s \Sigma / \partial t = \frac{12v}{s^2} \frac{\partial^2}{\partial s^2} (s \Sigma)$$

\therefore can write $s \Sigma = S(s) T(t)$ and

$$\frac{1}{T} dT/dt = \frac{12v}{s^2} \frac{1}{S} d^2 S / ds^2 = -\lambda^2 \quad (\text{ie have to be a constant to be indep of } t \text{ and } s)$$

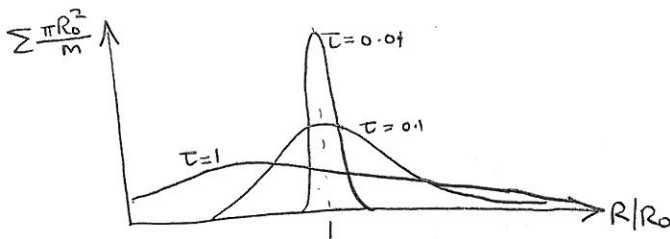
$\rightarrow T$ is exponential

S is a Bessel function

eg if $\Sigma(R, 0) = \frac{m}{2\pi R_0} \delta(R - R_0)$ (ie. start with all mass m at R_0)

then $\Sigma(r, t) = \left(\frac{m}{\pi R_0^2 t^{1/4}} \right) e^{-\frac{(1+x^2)}{t}} I_{1/4}(2x/t)$ - modified Bessel function

where $x = R/R_0$, $t = 12\nu t / R_0^2$ are dimensionless variables



- see part II QATAM project 23.8!
- most of the mass moves in, but outer parts move out (see ④)
- most ang mom ends up in small mass at large R.

- Note:
- characteristic spreading timescale is $t_{\text{visc}} \sim R_0^2 / \nu$ (also = momentum in annulus / torque on that annulus)
 - as Reynolds # $Re = R U_p / \nu$ $t_{\text{visc}} \sim \frac{R}{U_p} Re \sim t_{\text{orb}} Re$
 - for molecular viscosity $\nu \sim c_s l$ $\therefore Re = R U_p \sigma n / c_s$
 - \rightarrow as MMSN had $n \sim 10^{22} \text{ m}^{-3}$, $\sigma \sim 10^{-20} \text{ m}^2$, $T \sim 300 \text{ K} \rightarrow c_s \sim 1600 \text{ m/s}$
 - $\therefore Re \sim 10^{14}$ $(U_p \sim 3 \times 10^4 \text{ m/s})$
 - \rightarrow molecular viscosity is negligible
 - but flow is turbulent $\rightarrow \nu_{\text{turb}} \sim \nu_{\text{turb}} t_{\text{orb}}$ which can be significant (also MHD turbulence)

Solution for steady thin disk

Assume some supply of fluid maintaining steady $(\partial/\partial t)$ accretion rate \dot{m}

① $\rightarrow \frac{1}{R} \frac{\partial}{\partial R} [R \Sigma U_R] = 0$

$\therefore R \Sigma U_R = \text{const} = -\dot{m}/2\pi$

Ⓜ for Keplerian disk

$\rightarrow U_R = -\frac{\dot{m}}{2\pi} (R \Sigma)^{-1} = -3 \partial [v \Sigma R^{1/2}] / \partial R / \Sigma R^{1/2}$

$\therefore \frac{\dot{m}}{3\pi} [R^{1/2}] = [v \Sigma R^{1/2}]$

Use b.c. that $v \Sigma = 0$ at $R = R_{**}$ (since star rotation $\Omega_{**} < \Omega_K$ - and this is only way to achieve that)

$\therefore v \Sigma = \frac{\dot{m}}{3\pi} [1 - (R_{**}/R)^{1/2}]$

More realistically $\Omega \neq \Omega_K$ near star, but reaches a peak at $R = R_{**} + \epsilon$ & $\Omega_K(R_{**}) \pm \delta$ then decreases to Ω_{**} at star

But using ③, and integrating from $R_{**} + \epsilon$ to R , gives same result.

Consider rate of energy lost per unit area from an annulus due to viscous dissipation

$F_{\text{diss}} = \int \sigma_{ij} \partial_j u_i dV / 2\pi R dR$

(NB this was derived assuming fluid was incompressible, turns out to be OK as other terms are much smaller for thin disk)

$= \frac{1}{2} \int \eta (\partial_j u_i + \partial_i u_j)^2 dz$

$= v \Sigma (R d\Omega/dR)^2$

Using Kepler's law and above $v \Sigma$ eqn:

$F_{\text{diss}} = \frac{3GM\dot{m}}{4\pi R^3} [1 - (R_{**}/R)^{1/2}]$

If disk is optically thick, that energy is radiated away at a BB at temperature T_{eff} s.t:

as radiates from top and bottom of disk

$2 \int \sigma_{\text{SB}} T_{\text{eff}}^4(R) = F_{\text{diss}}$

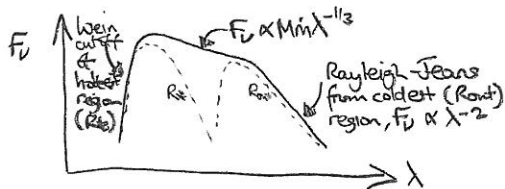
(ie. energy balance)

\therefore for $R \gg R_{**}$, $T_{\text{eff}} \propto R^{-3/4}$

And emission spectrum is

$F_\nu = \int_{R_{**}}^{R_{\text{out}}} B_\nu(T_{\text{eff}}(R)) 2\pi R dR$

where $B_\nu = \frac{2h}{c^2} \nu^3 / (e^{h\nu/kT} - 1) = \text{Planck function}$



NB this is independent of form of η and depends only on \dot{m} , R_{**} , R_{out} , M . So can't learn η from steady disks. (rather look at non steady disks, as changes to L occur on these timescales)

Disk luminosity $L = \int_{R_{**}}^{R_{\text{out}}} F_{\text{diss}} 2\pi R dR = GM\dot{m}/2R_{**}$

This energy comes from grav. pot-en. of infalling material, and since it loses GM/R_{**} by time it reaches the star $\rightarrow GM\dot{m}/R_{**}$ is rate of loss due to infall

\rightarrow half of this energy is radiated from disk (L)

\rightarrow other half is in KE of material as it reaches * (lost in boundary layer?)

Note that locally (ie. in an annulus) it is not true that grav. pot-en is converted into heat, since the energy eqn shows that this may be transferred to other annuli through viscous stresses (eq).

Magnetohydrodynamics (MHD)

49

Astrophysical fluids usually highly ionised (\therefore highly conducting) and permeated by magnetic fields

MHD is an approximation that considers interaction betw fluid and mag fields

\rightarrow particles moving through mag. field experience Lorentz force

(as flux is $\propto \underline{u} \wedge \underline{B}$ particles move freely along mag field lines but are constrained \perp)

\rightarrow particle motion also changes mag. field

Start with Maxwell's eqns

(M1) $\nabla \cdot \underline{B} = 0$ (no source or sink of mag. flux)

(M2) $\nabla \cdot \underline{E} = q / \epsilon_0$ (\underline{E} = electric field, q = charge density, $\epsilon_0 \stackrel{\text{(vacuum)}}{=} \text{electrical permittivity}$)

(M3) $\frac{1}{\mu_0} \nabla \wedge \underline{B} = \underline{j} + \epsilon_0 \partial \underline{E} / \partial t$ (μ_0 = magnetic permeability, \underline{j} = current density, 2nd term is displacement current)

(M4) $\partial \underline{B} / \partial t = -\nabla \wedge \underline{E}$ (electromagnetic induction)

Use Ohm's law to relate \underline{j} to \underline{E} and \underline{B} .

In the frame moving with fluid (dashes)

(O) $\underline{j}' = \sigma \underline{E}'$ (σ = electrical conductivity)

Use Lorentz transformation (see relativity course) for frame' wrt inertial frame (no dash)

$$\underline{E}' = (1-\gamma) \left(\frac{\underline{u} \cdot \underline{E}}{u^2} \right) \underline{u} + \gamma (\underline{E} + \underline{u} \wedge \underline{B})$$

$$\underline{B}' = (1-\gamma) \left(\frac{\underline{u} \cdot \underline{B}}{u^2} \right) \underline{u} + \gamma \left(\underline{B} - \frac{1}{c^2} \underline{u} \wedge \underline{E} \right)$$

where $\gamma = (1 - u^2/c^2)^{-1/2}$

Approximation (1) $u^2 \ll c^2$ (non relativistic)

\rightarrow Simplifies transformation to

$$\underline{E}' = \underline{E} + \underline{u} \wedge \underline{B}$$

$$\underline{B}' = \underline{B}$$

\rightarrow Can neglect displacement current st. (M3) $\Rightarrow \underline{j} = \frac{1}{\mu_0} \nabla \wedge \underline{B}$ (Ampere's law)

Why? Consider the flow has typical lengthscale L and timescale T , i.e. $u \sim L/T$

fractional contribution of d.c. $X_d = \mu_0 \epsilon_0 \partial \underline{E} / \partial t / \nabla \wedge \underline{B} \sim \mu_0 \epsilon_0 \left(\frac{E}{B} \right) \left(\frac{L}{T} \right)$

But $\mu_0 \epsilon_0 = \frac{1}{c^2}$ and (M4) $\Rightarrow E/B \sim L/T$

$$X_d \sim (u/c)^2$$

\rightarrow This in turn $\rightarrow \underline{j}' = \underline{j}$ (from the new (M3) is $\underline{B} = \underline{B}'$)

\rightarrow Ohm's law (O) $\Rightarrow \underline{j} = \sigma (\underline{E} + \underline{u} \wedge \underline{B})$

current induced by Lorentz forces

Magnetic flux

Equate (1) & (M3) $\Rightarrow \nabla \wedge \underline{B} = \mu_0 \sigma (\underline{E} + \underline{u} \wedge \underline{B})$
 Curl of flux $\Rightarrow \nabla \wedge (\nabla \wedge \underline{B}) = \mu_0 \sigma [\nabla \wedge \underline{E} + \nabla \wedge (\underline{u} \wedge \underline{B})]$
 Vector ID and (M4) $\Rightarrow -\nabla^2 \underline{B} + \nabla (\nabla \cdot \underline{B}) = \mu_0 \sigma [-\partial \underline{B} / \partial t + \nabla \wedge (\underline{u} \wedge \underline{B})]$
 Use (M1) $\therefore \underbrace{\partial \underline{B} / \partial t}_{\text{rate of change of mag field}} - \underbrace{\nabla \wedge (\underline{u} \wedge \underline{B})}_{\text{convection of field by fluid}} = \underbrace{\frac{1}{\mu_0 \sigma} \nabla^2 \underline{B}}_{\text{diffusion through conductive term}}$

Remember $\frac{D}{Dt} \int_S \underline{w} \cdot d\underline{S} = \int_S d\underline{S} \cdot [\partial \underline{w} / \partial t - \nabla \wedge (\underline{u} \wedge \underline{w})]$ applies to ANY quantity \underline{w}

So if $\Phi = \int_S \underline{B} \cdot d\underline{S}$ = magnetic flux through a surface

$D\Phi / Dt = \int_S d\underline{S} \cdot [\frac{1}{\mu_0 \sigma} \nabla^2 \underline{B}]$ (ie Φ only changes through diffusion term)

Approximation (2) $\sigma \rightarrow \infty$ (infinite conductivity)

$\Rightarrow D\Phi / Dt = 0$

\therefore Mag. field moves w fluid

A mag. field line consists of same particles at all times

The mag. field is "frozen in" to a perfectly conducting fluid

\rightarrow Ohm's law (1) $\Rightarrow \underline{E} + \underline{u} \wedge \underline{B} = 0$

\rightarrow Taking $\underline{B} \cdot$ this $\Rightarrow \underline{E} \cdot \underline{B} = 0$ and so \underline{E} is \perp to \underline{B}

\rightarrow (M4) becomes $\underline{\partial \underline{B} / \partial t} = \nabla \wedge (\underline{u} \wedge \underline{B})$ (induction eq.)

Lorentz force

Need to include the electromagnetic force acting on fluid in momentum eq. s.t.

(mom) $\rho (\partial \underline{u} / \partial t + \underline{u} \cdot \nabla \underline{u}) = -\nabla p + \underline{f}_L$

where $\underline{f}_L = q \underline{E} + \underline{j} \wedge \underline{B}$
 $= q \underline{E} + \frac{1}{\mu_0} (\nabla \wedge \underline{B}) \wedge \underline{B}$ (from (M3))

Approximation (3) $q \rightarrow 0$ (charge neutrality)

Consider fractional contribution of \underline{E} field to \underline{f}_L

$\chi_E = \frac{q \underline{E}}{\frac{1}{\mu_0} (\nabla \wedge \underline{B}) \wedge \underline{B}}$
 $= \frac{\mu_0 \sigma (\nabla \cdot \underline{E}) \underline{E}}{(\nabla \wedge \underline{B}) \wedge \underline{B}}$ (from (M2))
 $\approx (u/c)^2$ (ie. comes directly from non-relativistic assumption)

\rightarrow Set $q=0$ s.t. (M2) $\Rightarrow \nabla \cdot \underline{E} = 0$

and in (mom) $\underline{f}_L = \frac{1}{\mu_0} (\nabla \wedge \underline{B}) \wedge \underline{B}$

Magnetic pressure $= \frac{1}{\mu_0} [-\nabla (\frac{B^2}{2}) + (\underline{B} \cdot \nabla) \underline{B}]$

behaves like hydrostatic pressure of magnitude $p_{mag} = B^2 / 2\mu_0$

\therefore relative importance of ram vs thermal vs magnetic pressure dep on ratios of $\frac{1}{2} \rho u^2$ to ρc_s^2 to $B^2 / 2\mu_0$

and equating KE and mag. energy densities defines a velocity

$V_A = [B^2 / \rho \mu_0]^{1/2} = \text{Alfvén velocity}$

Charge Neutrality

Consider a static fluid composed of protons (+) and electrons (-)

Proton density is uniform $n^+ = n_0$

But there is small charge imbalance s.t. $n = n_0 + n_1(\mathbf{r}, t)$ where $n_1 \ll n_0$

Resulting charge density $q = n_0 e - (n_0 + n_1) e$ (where $e = \text{electron charge}$)
 $= -n_1 e$

So (M2) $\rightarrow \nabla \cdot \mathbf{E} = -n_1 e / \epsilon_0$

This results in a force on electrons causing a pert to their velocity $u_1(\mathbf{r}, t)$

$$\rightarrow m_e \frac{\partial u_1}{\partial t} = -e \mathbf{E}$$

Use continuity eqn

$$\rightarrow \frac{\partial n}{\partial t} + \nabla \cdot (n \mathbf{u}_1) = 0$$

$$\therefore \frac{\partial n_1}{\partial t} + n_0 \nabla \cdot \mathbf{u}_1 = 0 \quad (\text{to 1st order})$$

Differentiate $\rightarrow \frac{\partial^2 n_1}{\partial t^2} + n_0 \nabla \cdot \frac{\partial \mathbf{u}_1}{\partial t} = 0$

$$\therefore \frac{\partial^2 n_1}{\partial t^2} - n_0 \left(\frac{e}{m_e} \right) \nabla \cdot \mathbf{E} = 0 \quad (\text{sub in for } \frac{\partial \mathbf{u}_1}{\partial t})$$

$$\therefore \frac{\partial^2 n_1}{\partial t^2} + (n_0 e^2 / m_e \epsilon_0) n_1 = 0 \quad (\text{sub in for } \nabla \cdot \mathbf{E})$$

So charge imbalance oscillates at plasma frequency $\omega_p = \sqrt{n_0 e^2 / m_e \epsilon_0}$

$$\rightarrow \nu_p = \omega_p / 2\pi \approx 9.0 \sqrt{n_0} \text{ Hz}$$

eg for $n_0 \approx 10^{12} \text{ m}^{-3}$ in Earth's ionosphere $\nu_p \approx 10^7 \text{ Hz}$

Thus charge imbalances are rapidly neutralised

and more subtly \rightarrow EM rad in $\nu < \nu_p$ is reflected from ionosphere as electrons respond to incoming wave rather than allow it to be transmitted

If the electron velocity is the thermal velocity, ie $u_1 \approx \sqrt{\frac{kT_e}{m_e}}$
 then the lengthscale associated is oscillation's

$$\lambda_D = u_1 / \omega_p = \left(\epsilon_0 k T_e / n_0 e^2 \right)^{1/2} = \text{Debye length} = 70 \sqrt{T_e / n_0} \text{ m}$$

which is an "effective shielding length"

\rightarrow thermal motions iron out plasma oscillations on that scale

g. in ionosphere where $T_e \approx 10^3 \text{ K}$, $\lambda_D \approx 1 \text{ mm}$

Waves in plasmas

Ⓐ Equations: $\partial \rho / \partial t + \nabla \cdot (\rho \underline{u}) = 0$

$\underline{u} \cdot \nabla \underline{u} = -\frac{1}{\rho} \nabla p + \frac{1}{\rho} \frac{1}{\mu_0} (\nabla \wedge \underline{B}) \wedge \underline{B}$

$\frac{\partial \underline{B}}{\partial t} = \nabla \wedge (\underline{u} \wedge \underline{B})$
 $\underline{p} = K \rho^\gamma$

Ⓑ Unperturbed: $\rho = \rho_0, \underline{u} = \underline{u}_0 = 0, p = p_0, \underline{B} = \underline{B}_0$ ie uniform

Ⓒ Consider Lagrangian perturbation, called subscript 1

Since uniform medium $\delta \underline{x} = \underline{x}_1$

∴ substitute $\rho = \rho_0 + \rho_1, \underline{u} = \underline{u}_1, p = p_0 + p_1, \underline{B} = \underline{B}_0 + \underline{B}_1$

Ⓓ First order perturbed eqns

• $\frac{\partial \rho_1}{\partial t} + \rho_0 \nabla \cdot \underline{u}_1 = 0$

• $\rho_0 \frac{\partial \underline{u}_1}{\partial t} = -\nabla p_1 + \frac{1}{\mu_0} [\nabla \wedge (\underline{B}_0 + \underline{B}_1)] \wedge (\underline{B}_0 + \underline{B}_1)$
as uniform as 2nd order

∴ $\rho_0 \frac{\partial \underline{u}_1}{\partial t} + c_s^2 \nabla \rho_1 + \frac{1}{\mu_0} \underline{B}_0 \wedge (\nabla \wedge \underline{B}_1) = 0$

• $\frac{\partial \underline{B}_1}{\partial t} = \nabla \wedge (\underline{u}_1 \wedge \underline{B}_0)$

Ⓔ Wave eqn: $\partial / \partial t$ of mom eq.

$\rho_0 \frac{\partial^2 \underline{u}_1}{\partial t^2} + c_s^2 \nabla \frac{\partial \rho_1}{\partial t} + \frac{1}{\mu_0} \underline{B}_0 \wedge (\nabla \wedge \frac{\partial \underline{B}_1}{\partial t}) = 0$

$\frac{\partial^2 \underline{u}_1}{\partial t^2} - c_s^2 \nabla (\nabla \cdot \underline{u}_1) + \underline{v}_A \wedge (\nabla \wedge [\nabla \wedge (\underline{u}_1 \wedge \underline{v}_A)]) = 0$

where $\underline{v}_A = \underline{B}_0 / \sqrt{\mu_0 \rho_0} =$ Alfvén velocity (vectorial)

Ⓕ Solution: let $\underline{u}_1 = \underline{\tilde{u}}_1 e^{i(\underline{k} \cdot \underline{x} - \omega t)}$

$-\omega^2 \underline{u}_1 + c_s^2 (\underline{k} \cdot \underline{u}_1) \underline{k} + \underline{v}_A \wedge (\nabla \wedge [\nabla \wedge (\underline{u}_1 \wedge \underline{v}_A)]) = 0$

see handout: $-\omega^2 \underline{u}_1 + (c_s^2 + v_A^2) (\underline{k} \cdot \underline{u}_1) \underline{k} + (\underline{v}_A \cdot \underline{k}) [(\underline{v}_A \cdot \underline{k}) \underline{u}_1 - (\underline{v}_A \cdot \underline{u}_1) \underline{k} - (\underline{k} \cdot \underline{u}_1) \underline{v}_A] = 0$

\perp to lines of const phase
 \parallel to unperturbed field

If $\underline{k} \perp \underline{v}_A$ last term vanishes \rightarrow longitudinal magnetosonic wave

Phase velocity = $\sqrt{c_s^2 + v_A^2}$; ie depends on sum of hydrostatic and magnetic pressure!

sim to ordinary sound wave, but as field is frozen into fluid, mag field lines are bunched together in compressions \rightarrow extra magnetic pressure resisting compression



If $\underline{k} \parallel \underline{v}_A$

$\underline{k} = \frac{\underline{v}_A \underline{k}}{v_A}$, so $(k^2 v_A^2 - \omega^2) \underline{u}_1 + (\frac{c_s^2}{v_A^2} - 1) k^2 (\underline{v}_A \cdot \underline{u}_1) \underline{v}_A = 0$

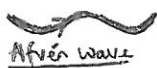
∴ Two types of motion

Ⓐ Longitudinal wave to $\underline{u}_1 \parallel \underline{k} \parallel \underline{v}_A \rightarrow \omega/k = c_s$ (as $(\underline{v}_A \cdot \underline{u}_1) \underline{v}_A = v_A^2 \underline{u}_1$)

∴ NB field is undisturbed by compressions, so no hydromagnetic forces!

∴ NB motion of fluid elements is transverse to dir of propagation

Ⓑ Transverse wave to $\underline{v}_A \cdot \underline{u}_1 = 0$ travelling at phase velocity $\omega/k = v_A$ (ie first term = 0)



This is purely magnetohydrodynamic wave that dep on tension in magnetic field lines and inertia of material which moves to field (since field is frozen in)

EGSun

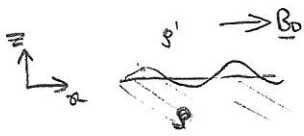
photosphere $\rho_0 \sim 10^4 \text{ kg/m}^3 \rightarrow v_A \approx 10^5 \text{ B m/s}$ whereas $T = 5770 \text{ K}$ so $c_s \approx 10^4 \text{ m/s}$
 at surface $B \sim 10^{-4} \text{ T}$ ($v_A \ll c_s$), but in sunspots $B \sim 0.3 \text{ T}$ ($v_A > c_s$)

EGMC

KE and mag E densities comparable and \gg thermal energies
 ∴ shocks softened if fluid elements collide at supersonic but sub Alfvénic speeds as Alfvén wave can carry news of impending coll to soften blow
 \rightarrow structure of dense DM req. understanding of magnetized shocks.

Rayleigh Taylor Instability in Magnetic fields

Can mag. fields stabilise fluids that are R-T unstable?



Consider 2 incompressible immiscible fluids in uniform (vertically down) grav. field w interface at $z=0$
 ie. as before but w/o flow in a dir. and apply uniform mag field $\parallel \hat{x}$ to interface

Equations: $\partial \rho / \partial t + \nabla \cdot (\rho \mathbf{u}) = 0$ (cont)
 $\rho (\partial \mathbf{u} / \partial t + \mathbf{u} \cdot \nabla \mathbf{u}) = -\nabla p + \rho \mathbf{g} + \frac{1}{\mu_0} (\nabla \wedge \mathbf{B}) \wedge \mathbf{B}$ (mom)
 $\partial \mathbf{B} / \partial t = \nabla \wedge (\mathbf{u} \wedge \mathbf{B})$ (M4)

where $\mathbf{g} = -g \hat{z}$

Equilibrium: $\mathbf{u}_0 = 0, \mathbf{B}_0 = B_0 \hat{x}, \rho = \rho_0, dp_0/dz = -\rho g$

Perturb (cont): $\xi = \xi(z) e^{i(kx - \omega t)}$ where $\xi(z) = [\xi_x(z), 0, \xi_z(z)]$
 $\mathbf{u} = \mathbf{u}_1 = \partial \xi / \partial t = -i \omega \xi$

(NB not looking at pert's in dir here but could do a more general analysis)

continuity with incompressibility $\rightarrow \nabla \cdot \mathbf{u} = 0$
 $\therefore ik \xi_x + d \xi_z / dz = 0$ *

Perturb (M4): $\mathbf{B} = \mathbf{B}_0 + \mathbf{B}_1(z) e^{i(kx - \omega t)}$ (NB this is Eulerian pert, note that B is frozen into fluid)
 so (M4) $\rightarrow -i \omega \mathbf{B}_1 = \nabla \wedge (-i \omega \xi \wedge \mathbf{B}_0)$ (to 1st order)
 $\therefore \mathbf{B}_1 = \nabla \wedge (\xi \wedge \mathbf{B}_0)$
 $= \nabla \wedge [0, B_0 \xi_x e^{i(kx - \omega t)}, 0]$
 $= [d \xi_x / dz, 0, ik \xi_x] B_0 e^{i(kx - \omega t)}$

So mag force term in perturbed mom. eq is (as $\nabla \wedge \mathbf{B}_0 = 0$)
 $\mathbf{f}_{mag} = \frac{1}{\mu_0} (\nabla \wedge \mathbf{B}_1) \wedge \mathbf{B}_0 = \frac{B_0^2}{\mu_0} [0, 0, d^2 \xi_x / dz^2 - k^2 \xi_x] e^{i(kx - \omega t)}$

Perturb (mom): incompressible $\rightarrow \Delta p = 0$
 but Eulerian pert is $p_1 = \rho_0 \phi - \xi_z \nabla p_0 = -\xi_z e^{i(kx - \omega t)} dp_0/dz$
 also define Eulerian pert p_1 st. $p = p_0 + p_1$ equilibrium
 so (mom) $\rightarrow (\rho_0 - \xi_z dp_0/dz) (-\omega^2 \xi + 2^{nd} \text{ order}) = -\nabla p_1 - (\rho_0 - \xi_z dp_0/dz) g \hat{z} + \frac{B_0^2}{\mu_0} (d^2 \xi_x / dz^2 - k^2 \xi_x) e^{i(kx - \omega t)}$
 $\square \xi_x \rightarrow -\omega^2 \rho_0 \xi_x e^{i(kx - \omega t)} = -dp_1/dz$
 $p_1 = \frac{\omega^2}{ik} \rho_0 \xi_x e^{i(kx - \omega t)}$ (integrate)
 $= -\frac{\omega^2}{(ik)^2} \rho_0 \frac{d \xi_x}{dz} e^{i(kx - \omega t)}$ (using *)
 $\therefore dp_1/dz = (\omega^2 / k) d[\rho_0 d \xi_x / dz] / dz e^{i(kx - \omega t)}$

$\square \xi_z \rightarrow -\omega^2 \rho_0 \xi_z = -(\omega^2 / k) d[\rho_0 d \xi_x / dz] / dz + \xi_z (dp_0/dz) g + \frac{B_0^2}{\mu_0} (d^2 \xi_x / dz^2 - k^2 \xi_x)$

Integrate this eqn w.r.t. z across interface apply b.c. that $\xi_x(z \rightarrow \pm \infty) \rightarrow 0$
 and that ξ_z is continuous across the interface at $z=0$

$\therefore \omega^2 = kg \left(\frac{\rho - \rho'}{\rho + \rho'} \right) + \frac{2}{\mu_0} \left(\frac{k^2 B_0^2}{\rho' + \rho} \right)$ this is $(k \cdot B_0)^2$ in more general case that oscilts is not necessarily in same dir as B_0

So, same result as before if $B_0 = 0$, (ie $\omega^2 < 0 \therefore$ unstable if $\rho' > \rho$)

But for $B_0 \neq 0$ mag fields have stabilising effect (as 2nd term is +ve) as work is expended in bending the field lines, $\bar{\omega}$ stronger effect for shorter wavelengths (as 2nd term $\propto k^2$)

For parts $\parallel k$ to B_0 there is critical k above which stable $k > \frac{g \mu_0}{2 B_0^2} (\rho' - \rho)$

But... \perp to B_0 there is no stabilising effect.